telephone (425) 227–1105; fax (425) 227–1320; email: *linh.le@faa.gov*.

SUPPLEMENTARY INFORMATION:

Background

On January 15, 2004, the Transport Airplane Directorate, Aircraft Certification Service, issued Policy Statement Number ANM–03–117–09, Policy Statement on Guidance for Determination of System, Hardware, and Software Development Assurance Levels on Transport Category Airplanes. This policy statement standardized the methodology for assigning development assurance levels (DAL) to systems, software, and complex electronic hardware.

In December 2010, the Society of Automotive Engineers (SAE) issued the document, Aerospace Recommended Practice (ARP) 4754A, Guidelines for Development of Civil Aircraft and Systems, as an acceptable method for establishing a development assurance process. This document contains an updated methodology for DAL assignment. In developing ARP 4754A, SAE considered the subject policy statement, with inputs and guidance from the FAA. In September 2011, the FAA issued AC 20-174, Development of Civil Aircraft and Systems, to formally recognize ARP 4754A and the DAL assignment methodology contained therein. The Transport Airplane Directorate determined that AC 20-174 and ARP 4754A provide an acceptable DAL assignment methodology for part 25 airplanes. To ensure correct understanding of the Transport Airplane Directorate's current policy for DAL assignments, the FAA is cancelling the subject policy, in favor of AC 20–174.

Cancellation of Policy Statement

As a result of the issuance of AC 20– 174, Policy Statement Number ANM– 03–117–09 is no longer in effect and is herewith cancelled.

Issued in Renton, Washington, on March 1, 2016.

Michael Kaszycki,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service, ANM–100.

[FR Doc. 2016–05529 Filed 3–10–16; 8:45 am] BILLING CODE 4910–13–P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

[Policy Statement No. PS-ANM100-1986-00051]

Policy Regarding Use of H–11 Bolts in Primary Structure on Transport Airplanes

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Notice of cancellation of policy statement.

SUMMARY: The Federal Aviation Administration (FAA) announces the cancellation of Policy Statement Number PS–ANM100–1986–00051. The policy statement is cancelled because it was superseded by an advisory circular (AC) and is no longer necessary.

DATES: This policy statement is cancelled on March 11, 2016.

FOR FURTHER INFORMATION CONTACT: Ian Won, Federal Aviation Administration, Transport Airplane Directorate, Transport Standards Staff, Airframe and Cabin Safety Branch, 1601 Lind Avenue SW., Renton, WA 98057–3356; telephone (425) 227–2145; fax (425) 227–1320; email: *Ian.Y.Won@faa.gov.*

SUPPLEMENTARY INFORMATION:

Background

On May 16, 1986, the Manager, Transport Airplane Directorate, Aircraft Certification Service, issued Policy Statement Number PS–ANM100–1986– 00051, *Policy Regarding Use of H–11 Bolts in Primary Structure on Transport Airplanes.* This policy statement addressed the adverse service history on H–11 bolts used in primary structure.

The FAA issued AC 20–127, Use of Society of Automotive Engineers (SAE) Class H11 Bolts, on July 8, 1987. The AC incorporates the same guidance as the older policy statement. The FAA should have cancelled the policy when they issued the AC but overlooked it.

Cancellation of Policy Statement

As a result of the issuance of AC 20– 127, Policy Statement Number PS– ANM100–1986–00051 is no longer in effect and is herewith cancelled.

Issued in Renton, Washington, on March 1, 2016.

Michael Kaszycki,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service, ANM–100.

[FR Doc. 2016–05532 Filed 3–10–16; 8:45 am] BILLING CODE 4910–13–P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

[Policy Statement No. PS-ANM100-1982-00038]

FAR 25.723 Energy Absorption Tests

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Notice of cancellation of policy statement.

SUMMARY: The Federal Aviation Administration (FAA) announces the cancellation of Policy Statement Number PS–ANM100–1982–00038. The policy statement is cancelled because it was superseded by an advisory circular (AC) and is no longer necessary.

DATES: This policy statement is cancelled on March 11, 2016.

FOR FURTHER INFORMATION CONTACT: Ian Won, Federal Aviation Administration, Transport Airplane Directorate, Transport Standards Staff, Airframe and Cabin Safety Branch, 1601 Lind Avenue SW., Renton, WA 98057–3356; telephone (425) 227–2145; fax (425) 227–1320; email: *Ian.Y.Won@faa.gov.*

SUPPLEMENTARY INFORMATION:

Background

On April 13, 1982, the Manager, Transport Airplane Directorate, Aircraft Certification Service, issued Policy Statement Number PS–ANM100–1982– 00038, *FAR 25.723 Energy Absorption Tests.* This policy statement addressed limits of extrapolation of landing gear drop test data.

The FAA issued AC 25.723–1, *Shock Absorption Tests,* on May 25, 2001, concurrently with Amendment 25–103, which changed the rule, 14 CFR 25.723. The AC addresses the same issue as the older policy statement but reflects the latest rule and is more specific. The FAA intended to cancel the policy when they released AC 25.723–1 but overlooked it.

Cancellation of Policy Statement

As a result of the issuance of AC 25.723–1 and Amendment 25–103, Policy Statement Number PS–ANM100–1982–00038 is no longer in effect and is herewith cancelled.

Issued in Renton, Washington, on March 1, 2016.

Michael Kaszycki,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service, ANM–100.

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