§ 23.1505 Airspeed limitations.

(a) The never-exceed speed \( V_{NE} \) must be established so that it is—
   (1) Not less than 0.9 times the minimum value of \( V_D \) allowed under § 23.335; and
   (2) Not more than the lesser of—
      (i) 0.9 \( V_D \) established under § 23.335; or
      (ii) 0.9 times the maximum speed shown under § 23.251.

(b) The maximum structural cruising speed \( V_{NO} \) must be established so that it is—
   (1) Not less than the minimum value of \( V_C \) allowed under § 23.335; and
   (2) Not more than the lesser of—
      (i) \( V_C \) established under § 23.335; or
      (ii) 0.89 \( V_{NE} \) established under paragraph (a) of this section.

(c) Paragraphs (a) and (b) of this section do not apply to turbine airplanes or to airplanes for which a design diving speed \( V_D/M_D \) is established under § 23.335(b)(4). For those airplanes, a maximum operating limit speed \( V_{MO}/M_{MO} \) must be established under paragraph (a) of this section.

§ 23.1507 Operating maneuvering speed.

The maximum operating maneuvering speed, \( V_O \), must be established as an operating limitation. \( V_O \) is a selected speed that is not greater than \( V_S/\sqrt{n} \) established in § 23.335(c).

§ 23.1511 Flap extended speed.

(a) The flap extended speed \( V_{FE} \) must be established so that it is—
   (1) Not less than the minimum value of \( V_F \) allowed in § 23.345(b); and
   (2) Not more than \( V_F \) established under § 23.345(a), (c), and (d).

(b) Additional combinations of flap setting, airspeed, and engine power may be established if the structure has been proven for the corresponding design conditions.

§ 23.1513 Minimum control speed.

The minimum control speed \( V_{MC} \), determined under § 23.149, must be established as an operating limitation.

§ 23.1519 Weight and center of gravity.

The weight and center of gravity limitations determined under § 23.23 must be established as operating limitations.

§ 23.1521 Powerplant limitations.

(a) General. The powerplant limitations prescribed in this section must be established so that they do not exceed the corresponding limits for which the engines or propellers are type certificated. In addition, other powerplant limitations used in determining compliance with this part must be established.

(b) Takeoff operation. The powerplant takeoff operation must be limited by—
   (1) The maximum rotational speed (rpm);
   (2) The maximum allowable manifold pressure (for reciprocating engines);
   (3) The maximum allowable gas temperature (for turbine engines);
   (4) The time limit for the use of the power or thrust corresponding to the limitations established in paragraphs (b)(1) through (3) of this section; and
   (5) The maximum allowable cylinder head (as applicable), liquid coolant and oil temperatures.

(c) Continuous operation. The continuous operation must be limited by—