

(f) 30-Second OEI power—thirty seconds at rated 30-Second OEI power.

(g) 2-Minute OEI power—two minutes at rated 2-Minute OEI power.

(h) Idle power—one minute at idle power.

§ 33.88 Engine Overtemperature Test.

In addition to the requirements of section 33.88, the following must be performed:

(a) For engines that do not provide a means for temperature limiting; conduct a test for a period of five minutes at the maximum permissible power-on RMP, with the gas temperature at least 75 degrees fahrenheit higher than the 30-Second OEI rating operating temperature limit.

(b) For engines that provide a means for temperature limiting; conduct a test for a period of four minutes at the maximum permissible power-on RPM, with the gas temperature at least 35 degrees fahrenheit higher than the 30-Second OEI rating operating temperature limit.

(c) A separate test engine may be used for each test.

(d) Following the test, rotor assembly growth and distress beyond serviceable limits for an overtemperature condition is permitted, provided the structural integrity of the rotor assembly is maintained, as shown by a procedure that is acceptable to the Administrator.

§ 33.93 Teardown Inspection.

In addition to the requirements of section 33.93, this special condition requires that the engine be completely disassembled after completing the additional testing of section 33.87. The engine must comply with section 33.93(a), but it may exhibit deterioration in excess of that permitted in section 33.93(b), and may include some engine parts and components that may be unsuitable for further use. It must be shown by procedures approved by the Administrator that the structural integrity of the engine, including mounts, cases, bearing supports, shafts and rotors, is maintained.

Issued in Burlington, Massachusetts, on November 16, 1995.

Jay J. Pardee,

*Manager, Engine & Propeller Directorate,
Aircraft Certification Service.*

[FR Doc. 95-28842 Filed 11-24-95; 8:45 am]

BILLING CODE 4910-13-M

14 CFR Part 39

[Docket No. 95-CE-01-AD; Amendment 39-9441; AD 95-24-11]

Airworthiness Directives; Fairchild Aircraft SA226 and SA227 Series Airplanes

AGENCY: Federal Aviation Administration, DOT.

ACTION: Final rule.

SUMMARY: This amendment adopts a new airworthiness directive (AD) that applies to certain Fairchild Aircraft SA226 and SA227 series airplanes. This action requires installing foreign object damage (FOD) barriers in the floorboards of the cockpit between the pedestal and floor from Fuselage Station (FS) 79.38 to FS 88.06 and on the outboard forward edge of the left-hand and right-hand cockpit forward floorboards at FS 79.38. Two incidents of objects falling through openings in the cockpit floor and jamming the elevator controls and the yoke prompted this action. The actions specified by this AD are intended to prevent airplane flight control jammings caused by objects falling through the cockpit floor openings.

DATES: Effective January 3, 1996. The incorporation by reference of certain publications listed in the regulations is approved by the Director of the Federal Register as of January 3, 1996.

ADDRESSES: Service information that applies to this AD may be obtained from Fairchild Aircraft, P.O. Box 790490, San Antonio, Texas 78279-0490; telephone (210) 824-9421. This information may also be examined at the Federal Aviation Administration (FAA), Central Region, Office of the Assistant Chief Counsel, Attention: Rules Docket 95-CE-01-AD, Room 1558, 601 E. 12th Street, Kansas City, Missouri 64106; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

FOR FURTHER INFORMATION CONTACT: Mr. Werner Koch, Aerospace Engineer, FAA, Airplane Certification Office, 2601 Meacham Boulevard, Fort Worth, Texas 76193-0150; telephone (817) 222-5133; facsimile (817) 222-5960.

SUPPLEMENTARY INFORMATION: A proposal to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) to include an AD that would apply to certain Fairchild Aircraft SA226 and SA227 series airplanes was published in the Federal Register on June 15, 1995 (60 FR 14235). The action proposed to require installing foreign object damage (FOD) barriers in the floorboards of the cockpit between the pedestal and floor

from Fuselage Station (FS) 79.38 to FS 88.06 and on the outboard forward edge of the left-hand and right-hand cockpit forward floorboards at FS 79.38.

Accomplishment of the proposed action would be in accordance with Fairchild Service Bulletin (SB) 226-53-012, Fairchild SB 227-53-005, or Fairchild SB CC7-53-002, all issued: September 22, 1994, as applicable.

Interested persons have been afforded an opportunity to participate in the making of this amendment. Due consideration has been given to the one comment received.

The commenter is in favor of the substance of the proposed rule, but feels that the FAA should have issued a final rule; request for comments, instead of a notice of proposed rulemaking (NPRM). Under current regulations, the FAA must issue an NPRM prior to issuing a final rule to allow the public the opportunity to comment, unless the FAA demonstrates that the unsafe condition is an urgent safety of flight condition. After reviewing all information related to this subject, the FAA made the determination prior to issuing the NPRM that the unsafe condition was not an urgent safety of flight condition, and thus did not require final rule; request for comments, AD action. The AD is unchanged as a result of this comment.

No comments were received on the FAA's determination of the cost to the public.

After careful review of all available information related to the subject presented above, the FAA has determined that air safety and the public interest require the adoption of the rule as proposed except for minor editorial corrections. The FAA has determined that these minor corrections will not change the meaning of the AD and will not add any additional burden upon the public than was already proposed.

The FAA estimates that 855 airplanes in the U.S. registry will be affected by this AD, that it will take approximately 4 workhours per airplane to accomplish the required action, and that the average labor rate is approximately \$60 an hour. Parts cost approximately \$50 per airplane. Based on these figures, the total cost impact of this AD on U.S. operators is estimated to be \$247,950. This figure is based on the assumption that no affected airplane owner/operator has incorporated the required modification and that parts have not been distributed to any owner/operator of the affected airplanes.

The regulations adopted herein will not have substantial direct effects on the States, on the relationship between the

national government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, in accordance with Executive Order 12612, it is determined that this final rule does not have sufficient federalism implications to warrant the preparation of a Federalism Assessment.

For the reasons discussed above, I certify that this action (1) is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A copy of the final evaluation prepared for this action is contained in the Rules Docket. A copy of it may be obtained by contacting the Rules Docket at the location provided under the caption **ADDRESSES**.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

Adoption of the Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration amends part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 USC 106(g), 40101, 40113, 44701.

§ Section 39.13 [AMENDED]

2. Section 39.13 is amended by adding a new airworthiness directive (AD) to read as follows:

95-24-11 Fairchild Aircraft: Amendment 39-9441; Docket No. 95-CE-01-AD.

Applicability: The following airplane models and serial numbers, certificated in any category:

Model	Serial Nos.
SA226-T	All serial numbers.
SA226-T(B) ...	All serial numbers.
SA226-AT	All serial numbers.
SA226-TC	All serial numbers.
SA227-AT	All serial numbers.
SA227-AC	All serial numbers.
SA227-BC	All serial numbers.
SA227-TT	All serial numbers.

Model	Serial Nos.
SA227-CC	CC784 and CC790 through CC863
SA227-DC	DC784 and DC790 through DC863

Note 1: This AD applies to each airplane identified in the preceding applicability provision, regardless of whether it has been modified, altered, or repaired in the area subject to the requirements of this AD. For airplanes that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (c) of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and, if the unsafe condition has not been eliminated, the request should include specific proposed actions to address it. Compliance: Required within the next 600 hours time-in-service after the effective date of this AD, unless already accomplished.

To prevent airplane flight control jammings caused by objects falling through the cockpit floor openings, accomplish the following:

(a) Install foreign object damage (FOD) barriers in the floorboards of the cockpit between the pedestal and floor from Fuselage Station (FS) 79.38 to FS 88.06 and on the outboard forward edge of the left-hand and right-hand cockpit forward floorboards at FS 79.38. Accomplish this action in accordance with the ACCOMPLISHMENT INSTRUCTIONS section of either Fairchild Service Bulletin (SB) 226-53-012, Fairchild SB 227-53-005, or Fairchild SB CC7-53-002, all Issued: September 22, 1994, as applicable.

(b) Special flight permits may be issued in accordance with sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be accomplished.

(c) An alternative method of compliance or adjustment of the compliance time that provides an equivalent level of safety may be approved by the Manager, Airplane Certification Office (ACO), FAA, 2601 Meacham Boulevard, Fort Worth, Texas 76193-0150. The request shall be forwarded through an appropriate FAA Maintenance Inspector, who may add comments and then send it to the Manager, Fort Worth ACO.

Note 2: Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the Fort Worth ACO.

(d) The installation required by this AD shall be done in accordance with Fairchild Service Bulletin 226-53-012, Fairchild Service Bulletin 227-53-005, or Fairchild Service Bulletin CC7-53-002, all Issued: September 22, 1994. This incorporation by reference was approved by the Director of the Federal Register in accordance with 5 U.S.C. 552(a) and 1 CFR part 51. Copies may be obtained from Fairchild Aircraft, P.O. Box 790490, San Antonio, Texas 78279-0490. Copies may be inspected at the FAA, Central

Region, Office of the Assistant Chief Counsel, Room 1558, 601 E. 12th Street, Kansas City, Missouri, or at the Office of the Federal Register, 800 North Capitol Street, NW., 7th Floor, suite 700, Washington, DC.

(e) This amendment (39-9441) becomes effective on January 3, 1996.

Issued in Kansas City, Missouri, on November 17, 1995.

Michael Gallagher,
Manager, Small Airplane Directorate, Aircraft Certification Service.

[FR Doc. 95-28794 Filed 11-24-95; 8:45 am]

BILLING CODE 4910-13-U

14 CFR Part 39

[Docket No. 95-NM-83-AD; Amendment 39-9434; AD 95-24-02]

Airworthiness Directives; Boeing Model 747SP Series Airplanes

AGENCY: Federal Aviation Administration, DOT.

ACTION: Final rule.

SUMMARY: This amendment adopts a new airworthiness directive (AD), applicable to certain Boeing Model 747SP series airplanes, that requires modification of the escape slide/raft on Door 2 of the airplane. This amendment is prompted by reports indicating that the escape slide/raft on Door 2 deployed onto the wing of the airplane and did not inflate automatically. The actions specified by this AD are intended to ensure that the escape slide/raft on Door 2 inflates automatically so that passengers are able to exit the airplane through Door 2 in the event of an emergency evacuation.

DATES: Effective December 27, 1995.

The incorporation by reference of certain publications listed in the regulations is approved by the Director of the Federal Register as of December 27, 1995.

ADDRESSES: The service information referenced in this AD may be obtained from BFGoodrich Company, Aircraft Evacuation Systems, Department 7916, Phoenix, Arizona 85040. This information may be examined at the Federal Aviation Administration (FAA), Transport Airplane Directorate, Rules Docket, 1601 Lind Avenue, SW., Renton, Washington; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

FOR FURTHER INFORMATION CONTACT: Monica Nemecek, Aerospace Engineer, Airframe Branch, ANM-120S, FAA, Transport Airplane Directorate, Seattle Aircraft Certification Office, 1601 Lind Avenue, SW., Renton, Washington 98055-4056; telephone (206) 227-2773; fax (206) 227-1181.