

location provided under the caption
ADDRESSES.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Safety.

The Proposed Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration proposes to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

2. Section 39.13 is amended by adding the following new airworthiness directive:

Boeing: Docket 98–NM–58–AD.

Applicability: Model 737–100, –200, –300, –400, and –500 series airplanes; as listed in Boeing Alert Service Bulletin 737–53A1173, Revision 2, dated January 15, 1998; certificated in any category.

Note 1: This AD applies to each airplane identified in the preceding applicability provision, regardless of whether it has been modified, altered, or repaired in the area subject to the requirements of this AD. For airplanes that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (d) of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and, if the unsafe condition has not been eliminated, the request should include specific proposed actions to address it.

Compliance: Required as indicated, unless accomplished previously.

To prevent fatigue cracking of the forward pressure bulkhead, which could result in rapid decompression of the airplane fuselage, accomplish the following:

(a) Perform inspections of the center web, vertical chords, and side chord areas of the forward pressure bulkhead for fatigue cracking, in accordance with the Accomplishment Instructions of Boeing Alert Service Bulletin 737–53A1173, Revision 2, dated January 15, 1998, at the time specified in paragraph (a)(1) or (a)(2) of this AD, as applicable. Thereafter, repeat the inspections at intervals not to exceed 3,000 flight cycles until the preventive modifications required by paragraph (d) of this AD have been accomplished.

(1) For airplanes that have accumulated 60,000 or more total flight cycles as of the effective date of this AD: Inspect within 1,500 flight cycles after the effective date of this AD.

(2) For airplanes that have accumulated fewer than 60,000 total flight cycles as of the effective date of this AD: Inspect prior to the accumulation of 15,000 total flight cycles, or within 3,000 flight cycles after the effective date of this AD, whichever occurs later.

(b) If any crack is found during any inspection required by paragraph (a) of this AD, prior to further flight, repair the area in accordance with the Accomplishment Instructions of Boeing Alert Service Bulletin 737–53A1173, Revision 2, dated January 15, 1998; except, where the alert service bulletin specifies that the manufacturer may be contacted for repair instructions, repair in accordance with a method approved by the Manager, Seattle Aircraft Certification Office (ACO), FAA, Transport Airplane Directorate.

(c) Prior to the accumulation of 75,000 total flight cycles, or within 12,000 flight cycles after the effective date of this AD, whichever occurs later: Accomplish preventive modifications of the center web, vertical chords, and side chord areas of the forward pressure bulkhead, in accordance with the Accomplishment Instructions of Boeing Alert Service Bulletin 737–53A1173, Revision 2, dated January 15, 1998. Accomplishment of these modifications constitutes terminating action for the inspections required by paragraph (a) of this AD, except for the requirement to inspect the side chord areas at water line 207 (for which no preventive modification is described in the alert service bulletin). For these side chord areas, continue inspecting in accordance with the requirements of paragraph (a) of this AD.

(d) An alternative method of compliance or adjustment of the compliance time that provides an acceptable level of safety may be used if approved by the Manager, Seattle ACO. Operators shall submit their requests through an appropriate FAA Principal Maintenance Inspector, who may add comments and then send it to the Manager, Seattle ACO.

Note 2: Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the Seattle ACO.

(e) Special flight permits may be issued in accordance with sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be accomplished.

Issued in Renton, Washington, on September 25, 1998.

Darrell M. Pederson,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.
[FR Doc. 98–27124 Filed 10–8–98; 8:45 am]

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 96–CE–60–AD]

RIN 2120–AA64

Airworthiness Directives; Raytheon Aircraft Company Models 1900, 1900C, and 1900D Airplanes

AGENCY: Federal Aviation Administration, DOT.

ACTION: Notice of proposed rulemaking (NPRM).

SUMMARY: This document proposes to revise Airworthiness Directive (AD) 97–15–13 R1, which currently requires installing lubrication fittings in the airstair door handle and latch housing mechanisms on certain Raytheon Aircraft Company (Raytheon) Models 1900, 1900C, and 1900D airplanes (commonly referred to as Beech Models 1900, 1900C, and 1900D airplanes). Since issuance of AD 97–15–13 R1, Raytheon has revised the applicable service information to correct the reference to the number of parts each owner/operator of the affected airplanes should order and to change an incorrect reference to a maintenance manual. The proposed AD would retain the actions of AD 97–15–13 R1, and would incorporate the revised service bulletin into the proposed AD. The actions specified by the proposed AD are intended to continue to prevent moisture from accumulating and freezing in the airstair door handle and latch housing, which could result in the door freezing shut and passengers not being able to evacuate the airplane in an emergency situation.

DATES: Comments must be received on or before December 19, 1998.

ADDRESSES: Submit comments in triplicate to the Federal Aviation Administration (FAA), Central Region, Office of the Regional Counsel, Attention: Rules Docket No. 96–CE–60–AD, Room 1558, 601 E. 12th Street, Kansas City, Missouri 64106. Comments may be inspected at this location between 8 a.m. and 4 p.m., Monday through Friday, holidays excepted.

Service information that applies to the proposed AD may be obtained from the Raytheon Aircraft Company, P.O. Box 85, Wichita, Kansas 67201–0085. This information also may be examined at the Rules Docket at the address above.

FOR FURTHER INFORMATION CONTACT: Mr. Steven E. Potter, Aerospace Safety Engineer, FAA, Wichita Aircraft Certification Office, 1801 Airport Road,

Mid-Continent Airport, Wichita, Kansas 67209; telephone: (316) 946-4124; facsimile: (316) 946-4407.

SUPPLEMENTARY INFORMATION:

Comments Invited

Interested persons are invited to participate in the making of the proposed rule by submitting such written data, views, or arguments as they may desire. Communications should identify the Rules Docket number and be submitted in triplicate to the address specified above. All communications received on or before the closing date for comments, specified above, will be considered before taking action on the proposed rule. The proposals contained in this notice may be changed in light of the comments received.

Comments are specifically invited on the overall regulatory, economic, environmental, and energy aspects of the proposed rule. All comments submitted will be available, both before and after the closing date for comments, in the Rules Docket for examination by interested persons. A report that summarizes each FAA-public contact concerned with the substance of this proposal will be filed in the Rules Docket.

Commenters wishing the FAA to acknowledge receipt of their comments submitted in response to this notice must submit a self-addressed, stamped postcard on which the following statement is made: "Comments to Docket No. 96-CE-60-AD." The postcard will be date stamped and returned to the commenter.

Availability of NPRMs

Any person may obtain a copy of this NPRM by submitting a request to the FAA, Central Region, Office of the Regional Counsel, Attention: Rules Docket No. 96-CE-60-AD, Room 1558, 601 E. 12th Street, Kansas City, Missouri 64106.

Discussion

AD 97-15-13 R1, Amendment 39-10131 (62 FR 49426, September 22, 1997), currently requires installing lubrication fittings in the airstair door handle and latch housing mechanisms on certain Raytheon Models 1900, 1900C, and 1900D airplanes. Accomplishment of these actions are required in accordance with Raytheon Mandatory Service Bulletin No. 2572, Issued: July, 1996.

The actions specified by AD 97-15-13 R1 are intended to prevent moisture from accumulating and freezing in the airstair door handle and latch housing, which could result in the door freezing

shut and passengers not being able to evacuate the airplane in an emergency situation.

AD 97-15-13 R1 was the result of reports of the airstair door not opening because the door was frozen shut on the above-referenced airplanes.

Actions Since Issuance of Previous Rule

Since AD 97-15-13 R1 has become effective, Raytheon has issued Mandatory Service Bulletin SB.2572, Issued: July, 1996; Revision No. 1, May, 1998. This service bulletin revision corrects the reference to the number of parts each owner/operator of the affected airplanes should order and changes an incorrect reference to a maintenance manual.

The FAA's Determination

After examining the circumstances and reviewing all available information related to the subject matter described above, the FAA has determined that: —Raytheon Mandatory Service Bulletin SB.2572, Issued: July, 1996; Revision No. 1, May, 1998, should be incorporated into AD 97-15-13 R1; and —AD action should be taken to continue to prevent moisture from accumulating and freezing in the airstair door handle and latch housing, which could result in the door freezing shut and passengers not being able to evacuate the airplane in an emergency situation.

Explanation of the Provisions of the Proposed AD

Since an unsafe condition has been identified that is likely to exist or develop in other Raytheon Models 1900, 1900C, and 1900D airplanes of the same type design, the proposed AD would revise AD 97-15-13 R1. The proposed AD would retain the requirements in AD 97-15-13 R1 of installing lubrication fittings in the airstair door handle and latch housing mechanisms. Accomplishment of the proposed installations would be required in accordance with Raytheon Mandatory Service Bulletin No. 2572, Issued: July, 1996; or Raytheon Mandatory Service Bulletin SB.2572, Issued: July, 1996; Revision No. 1, May, 1998.

Cost Impact

The FAA estimates that 408 airplanes in the U.S. registry would be affected by the proposed AD, that it would take approximately 14 workhours per airplane to accomplish the proposed installation, and that the average labor rate is approximately \$60 an hour. Parts cost approximately \$50 per airplane. Based on these figures, the total cost

impact of the proposed AD on U.S. operators is estimated to be \$363,120, or \$890 per airplane. This figure is based on the presumption that no owner/operator of the affected airplanes has accomplished the required installation.

The proposed AD would require the same actions as AD 97-15-13 R1. The only difference is reference to Raytheon Mandatory Service Bulletin SB.2572, Issued: July, 1996; Revision No. 1, May, 1998. Therefore, the proposed AD imposes no additional cost impact upon U.S. owners/operators of the affected airplanes than is already required by AD 97-15-13 R1.

Regulatory Impact

The regulations proposed herein would not have substantial direct effects on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, in accordance with Executive Order 12612, it is determined that this proposal would not have sufficient federalism implications to warrant the preparation of a Federalism Assessment.

For the reasons discussed above, I certify that this action (1) is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) if promulgated, will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A copy of the draft regulatory evaluation prepared for this action has been placed in the Rules Docket. A copy of it may be obtained by contacting the Rules Docket at the location provided under the caption ADDRESSES.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Safety.

The Proposed Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration proposes to amend 14 CFR part 39 of the Federal Aviation Regulations as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

2. Section 39.13 is amended by removing Airworthiness Directive (AD) 97-15-13 R1, Amendment 39-10131, and by adding a new AD to read as follows:

Raytheon Aircraft Company (Type

Certificate No. A24CE formerly held by the Beech Aircraft Corporation): Docket No. 96-CE-60-AD; Revises AD 97-15-13 R1, Amendment 39-10131.

Applicability: The following airplane models and serial numbers, certificated in any category:

Model	Serial Nos.
1900	UA-1 through UA-3.
1900C	UB-1 through UB-74, and UC-1 through UC-174.
1900C (C-12J)	UD-1 through UD-6.
1900D	UE-1 through UE-157.

Note 1: This AD applies to each airplane identified in the preceding applicability provision, regardless of whether it has been modified, altered, or repaired in the area subject to the requirements of this AD. For airplanes that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (d) of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and, if the unsafe condition has not been eliminated, the request should include specific proposed actions to address it.

Compliance: Required within the next 200 hours time-in-service after September 27, 1997 (the effective date of AD 97-15-13 R1), unless already accomplished.

To prevent moisture from accumulating and freezing in the airstair door handle and latch housing, which could result in the door freezing shut and passengers not being able to evacuate the airplane in an emergency situation, accomplish the following:

(a) Install lubrication fittings in the airstair door handle and latch housing mechanisms in accordance with the ACCOMPLISHMENT INSTRUCTIONS section of either:

(1) Raytheon Mandatory Service Bulletin No. 2572, Issued: July, 1996; or

(2) Raytheon Mandatory Service Bulletin SB.2572, Issued: July, 1996; Revision No. 1, May, 1998.

(b) Special flight permits may be issued in accordance with sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be accomplished.

(c) An alternative method of compliance or adjustment of the compliance time that provides an equivalent level of safety may be approved by the Manager, Wichita Aircraft Certification Office (ACO), 1801 Airport Road, Room 100, Mid-Continent Airport, Wichita, Kansas 67209.

(1) The request shall be forwarded through an appropriate FAA Maintenance Inspector,

who may add comments and then send it to the Manager, Wichita ACO.

(2) Alternative methods of compliance approved in accordance with AD 97-15-13 R1 are considered approved as alternative methods of compliance for this AD.

Note 2: Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the Wichita ACO.

(d) All persons affected by this directive may obtain copies of the documents referred to herein upon request to the Raytheon Aircraft Company, P.O. Box 85, Wichita, Kansas 67201-0085; or may examine these documents at the FAA, Central Region, Office of the Regional Counsel, Room 1558, 601 E. 12th Street, Kansas City, Missouri 64106.

(e) This amendment revises AD 97-15-13 R1, Amendment 39-10131.

Issued in Kansas City, Missouri, on October 1, 1998.

Michael Gallagher,

Manager, Small Airplane Directorate, Aircraft Certification Service.

[FR Doc. 98-27122 Filed 10-8-98; 8:45 am]

BILLING CODE 4910-13-U

DEPARTMENT OF TRANSPORTATION**Federal Aviation Administration****14 CFR Part 39**

[Docket No. 98-CE-89-AD]

RIN 2120-AA64

Airworthiness Directives; Twin Commander Aircraft Corporation Model 680FL Airplanes

AGENCY: Federal Aviation Administration, DOT.

ACTION: Notice of proposed rulemaking (NPRM).

SUMMARY: This document proposes to adopt a new airworthiness directive (AD) that would apply to Twin Commander Aircraft Corporation (Twin Commander) Model 680FL airplanes. The proposed AD would require revising the FAA-approved Airplane Flight Manual (AFM) to specify procedures that would prohibit flight in severe icing conditions (as determined by certain visual cues), limit or prohibit the use of various flight control devices while in severe icing conditions, and provide the flight crew with recognition cues for, and procedures for exiting from, severe icing conditions. The proposed AD is prompted by the results of a review of the requirements for certification of these airplanes in icing conditions, new information on the icing environment, and icing data provided currently to the flight crew. The actions specified by the proposed AD are intended to minimize the

potential hazards associated with operating these airplanes in severe icing conditions by providing more clearly defined procedures and limitations associated with such conditions.

DATES: Comments must be received on or before December 2, 1998.

ADDRESSES: Submit comments in triplicate to the Federal Aviation Administration (FAA), Central Region, Office of the Regional Counsel, Attention: Rules Docket No. 98-CE-89-AD, Room 1558, 601 E. 12th Street, Kansas City, Missouri 64106. Comments may be inspected at this location between 8 a.m. and 4 p.m., Monday through Friday, holidays excepted.

FOR FURTHER INFORMATION CONTACT: Mr. John P. Dow, Sr., Aerospace Engineer, FAA, Small Airplane Directorate, 1201 Walnut, suite 900, Kansas City, Missouri 64106; telephone: (816) 426-6932; facsimile: (816) 426-2169.

SUPPLEMENTARY INFORMATION:

Comments Invited

Interested persons are invited to participate in the making of the proposed rule by submitting such written data, views, or arguments as they may desire. Communications should identify the Rules Docket number and be submitted in triplicate to the address specified above. All communications received on or before the closing date for comments, specified above, will be considered before taking action on the proposed rule. The proposals contained in this notice may be changed in light of the comments received.

Comments are specifically invited on the overall regulatory, economic, environmental, and energy aspects of the proposed rule. All comments submitted will be available, both before and after the closing date for comments, in the Rules Docket for examination by interested persons. A report that summarizes each FAA-public contact concerned with the substance of this proposal will be filed in the Rules Docket.

Commenters wishing the FAA to acknowledge receipt of their comments submitted in response to this notice must submit a self-addressed, stamped postcard on which the following statement is made: "Comments to Docket No. 98-CE-89-AD." The postcard will be date stamped and returned to the commenter.

Availability of NPRMs

Any person may obtain a copy of this NPRM by submitting a request to the FAA, Central Region, Office of the Regional Counsel, Attention: Rules