associations, has assets and annual income in excess of the amounts that would qualify them as small entities. Therefore, Farm Credit System institutions are not "small entities" as defined in the Regulatory Flexibility Act.

List of Subjects in 12 CFR Part 614

Agriculture, Banks, banking, Foreign trade, Reporting and recordkeeping requirements, Rural areas.

Accordingly, for the reasons stated in the preamble, part 614 of chapter VI, title 12 of the Code of Federal Regulations, is proposed to be amended as follows:

PART 614—LOAN POLICIES AND OPERATIONS

1. The authority citation for part 614 continues to read as follows:

Authority: 42 U.S.C. 4012a, 4104a, 4104b, 4106, and 4128; secs. 1.3, 1.5, 1.6, 1.7, 1.9, 1.10, 1.11, 2.0, 2.2, 2.3, 2.4, 2.10, 2.12, 2.13, 2.15, 3.0, 3.1, 3.3, 3.7, 3.8, 3.10, 3.20, 3.28, 4.12, 4.12A, 4.13B, 4.14, 4.14A, 4.14C, 4.14D, 4.14E, 4.18, 4.18A, 4.19, 4.25, 4.26, 4.27, 4.28, 4.36, 4.37, 5.9, 5.10, 5.17, 7.0, 7.2, 7.6, 7.8, 7.12, 7.13, 8.0, 8.5 of the Farm Credit Act (12 U.S.C. 2011, 2013, 2014, 2015, 2017, 2018, 2019, 2071, 2073, 2074, 2075, 2091, 2093, 2094, 2097, 2121, 2122, 2124, 2128, 2129, 2131, 2141, 2149, 2183, 2184, 2201, 2202, 2202a, 2202c, 2202d, 2202e, 2206, 2206a, 2207, 2211, 2212, 2213, 2214, 2219a, 2219b, 2243, 2244, 2252, 2279a, 2279a-2, 2279b, 2279c-1, 2279f, 2279f-1, 2279aa, 2279aa-5); sec. 413 of Pub. L. 100-233, 101 Stat. 1568, 1639.

Subpart B—Chartered Territories

2. Amend § 614.4070 by adding a new paragraph (d) to read as follows:

§614.4070 Loans and chartered territory— Farm Credit Banks, agricultural credit banks, Federal land bank associations, Federal land credit associations, production credit associations, and agricultural credit associations.

* * * *

(d) A bank or association chartered under title I or II of the Act may finance eligible borrower operations conducted wholly or partially outside its chartered territory through the purchase of loans from the Federal Deposit Insurance Corporation in compliance with § 614.4325(b)(3), provided:

(1) Notice is given to the Farm Credit System institution(s) chartered to serve the territory where the headquarters of borrower's operation being financed is located; and

(2) After loan purchase, additional financing of eligible borrower operations complies with paragraphs(a), (b), and (c) of this section.

Subpart H—Loan Purchases and Sales

3. Amend § 614.4325 by revising paragraph (b) to read as follows:

§614.4325 Purchase and sale of interests in loans.

(b) Authority to purchase and sell interests in loans. Loans and interests in loans may only be sold in accordance with each institution's lending authorities, as set forth in subpart A of this part. No Farm Credit System institution may purchase any interest in a loan from an institution that is not a Farm Credit System institution, except:

(1) For the purpose of pooling and securitizing such loans under title VIII of the Act;

(2) Purchases of a participation interest that qualifies under the institution's lending authority, as set forth in subpart A of this part and meets the requirements of § 614.4330 of this subpart;

(3) Loans purchased from the Federal Deposit Insurance Corporation, provided that the Farm Credit System institution with direct lending authority under titles I, II, or III of the Act:

(i) Conducts reasonable due diligence prior to purchase, and conducts thorough review after purchase, to determine that the loan, or pool of loans, qualifies under the institution's lending authority as set forth in subpart A of this part, and meets scope of financing and eligibility requirements in subpart A or subpart B of part 613;

(ii) Obtains funding bank approval, if a Farm Credit System association, for loans or pools of loans purchased exceeding 10 percent of total capital;

(iii) Establishes a program whereby each eligible borrower of the loan purchased is offered an opportunity to acquire the institution's required minimum amount of voting stock;

(iv) Determines whether each loan purchased, except for loans purchased that could be financed only by a bank for cooperatives under title III of the Act, is a distressed loan as defined in \S 617.7000, and provides the borrower of the purchased loan the rights afforded in \S 617.7000, subparts A, and D through G if the loan is distressed regardless of whether the loan is to an eligible or ineligible borrower; and

(v) Divests itself of ineligible loans purchased that are not distressed loans as defined in § 617.7000 and purchased loans of borrowers who elect not to acquire stock under the program offered in paragraph (b)(3)(iii) of this section in the same manner it would divest, under its current business practices, a loan in its loan portfolio determined to be ineligible.

* * *

Dated: May 12, 2010.

Roland E. Smith,

Secretary, Farm Credit Administration Board. [FR Doc. 2010–11772 Filed 5–17–10; 8:45 am] BILLING CODE 6705–01–P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 25

[Docket No. NM428; Notice No. 25–99–11– SC]

Special Conditions: Boeing 747–468, Installation of a Medical Lift

AGENCY: Federal Aviation Administration (FAA), DOT. **ACTION:** Notice of proposed special

conditions.

SUMMARY: This action proposes special conditions for the Boeing 747–468 airplane. This airplane, as modified by Jet Aviation, will have a novel or unusual design feature associated with the installation of a medical lift. The applicable airworthiness regulations do not contain adequate or appropriate safety standards for this design feature. These proposed special conditions contain the additional safety standards that the Administrator considers necessary to establish a level of safety equivalent to that established by the existing airworthiness standards.

DATES: We must receive your comments by June 17, 2010.

ADDRESSES: You must mail two copies of your comments to: Federal Aviation Administration, Transport Airplane Directorate, Attn: Rules Docket (ANM– 113), Docket No. NM428, 1601 Lind Avenue, SW., Renton, Washington 98057–3356. You may deliver two copies to the Transport Airplane Directorate at the above address. You must mark your comments: Docket No. NM428. You can inspect comments in the Rules Docket weekdays, except Federal holidays, between 7:30 a.m. and 4 p.m.

FOR FURTHER INFORMATION CONTACT: Jayson Claar, Transport Airplane Directorate, Aircraft Certification Service, 1601 Lind Avenue, SW., Renton, Washington 98057–3356; telephone (425) 227–2194; facsimile (425) 227–1149.

SUPPLEMENTARY INFORMATION:

Comments Invited

We invite interested people to take part in this rulemaking by sending written comments, data, or views. The most helpful comments reference a specific portion of the special conditions, explain the reason for any recommended change, and include supporting data. We ask that you send us two copies of written comments.

We will file in the docket all comments we receive, as well as a report summarizing each substantive public contact with FAA personnel concerning these special conditions. You can inspect the docket before and after the comment closing date. If you wish to review the docket in person, go to the address in the **ADDRESSES** section of this preamble between 7:30 a.m. and 4 p.m., Monday through Friday, except Federal holidays.

We will consider all comments we receive on or before the closing date for comments. We will consider comments filed late if it is possible to do so without incurring expense or delay. We may change these special conditions based on the comments we receive.

If you want us to acknowledge receipt of your comments on this proposal, include with your comments a selfaddressed, stamped postcard on which you have written the docket number. We will stamp the date on the postcard and mail it back to you.

Background

On March 2, 2007, Jet Aviation Engineering Services L.P. (JAES), of Teterboro, New Jersey, applied for a supplemental type certificate for a reconfiguration of an aircraft interior in a 747–468. The Boeing Model 747–468 airplane is FAA approved under Type Certificate A20WE as a large transportcategory airplane that is limited to 660 passengers or fewer, depending on the interior configuration.

This modification includes the installation of a medical lift between the main deck and upper deck. The lift allows the transport of a single occupant between the decks during cruise or ramp operations. The applicable airworthiness regulations do not contain adequate or appropriate safety standards for this design feature.

Type Certification Basis

Under the provisions of Title 14, Code of Federal Regulations (14 CFR) 21.101, JAES must show that the 747–468, as changed, continues to meet the applicable provisions of the regulations incorporated by reference in Type Certificate A20WE, or of the applicable regulations in effect on the date of application for the change. The regulations incorporated by reference in the type certificate are commonly referred to as the "original typecertification basis." The regulations incorporated by reference in Type Certificate A20WE are as follows:

• Part 36, as amended by Amendments 36–1 through 36–15, and any later amendments in existence at the time of certification.

• Special Federal Aviation Regulation (SFAR) 27, as amended by Amendments 27–1 through 27–6 and any later amendments in existence at the time of type certification.

• Part 25, effective February 1, 1965, as amended by Amendments 25–1 through 25–59, and the part 25 section-number exceptions itemized in Type Certificate A20WE.

The following special conditions, exemptions, and equivalent safety findings, which are part of the Model 747–300 certification basis, are also part of the certification basis for the Model 747–400.

The special conditions include those enclosed with an FAA letter to The Boeing Company dated February 20, 1970, and the following:

1. Special Condition 4A, revised to apply to airplanes with the landing-gear load-evener system deleted, was recorded as an enclosure to an FAA letter to The Boeing Company dated May 12, 1971.

2. Special Condition No. 25–61–NW– 1, for occupancy not to exceed 32 passengers on the upper deck of airplanes with a spiral staircase, was transmitted to The Boeing Company by FAA letter dated February 26, 1975.

3. Special Condition No. 25–71–NW– 3, for occupancy not to exceed 45 passengers on the upper deck of airplanes with a straight-segmented stairway, was transmitted to The Boeing Company by FAA letter dated September 8, 1976.

4. Modification of Special Condition No. 25–71–NW–3, for occupancy not to exceed 110 passengers on the upper deck of airplanes with a straightsegmented stairway, was transmitted to The Boeing Company by FAA letter dated August 3, 1981.

5. Special Condition No. 25–77–NW– 4, modification of the autopilot system to approve the airplane for use of the system under Category IIIb landing conditions, was transmitted to The Boeing Company by FAA letter dated July 8, 1977.

6. Special Condition No. 25–ANM–16, for use of an overhead crew-rest area, occupancy not to exceed ten crewmembers, was transmitted to The Boeing Company by FAA letter dated November 19, 1987. The FAA-approved procedures required for compliance with paragraph 13 of the special condition are located in Boeing Document D926U303, Appendix D.

7. Special Condition No. 25–ANM–24, applicable to flight-deck displays and propulsion-control systems, was provided to Boeing on December 22, 1988.

8. Special Condition No. 25–ANM–25, which established lightning-and radiofrequency-energy protection requirements, was provided to Boeing on December 22, 1988.

Exemptions From Part 25

Exemption no. 1013A, dated December 24, 1969: Exemption from Section 25.471(b) to allow lateral displacement of the center of gravity from the airplane centerline.

The following optional requirements, which are part of the Model 747–300 certification basis, apply also to the 747–400:

Requirement	Section
Ditching provisions	25.801
Ice-protection provisions	25.1419

The following equivalent-safety findings, previously made for earlier models under the provisions of § 21.21(b)(1), are also applicable to the Model 747–400:

Width of aisle 25.815. Pilot-compartment view 25.773. Use of 1-g stall speed (nonstructural items) Several (7 Use of 1-g stall speed (structural items) Several (7 Position-light distribution and intensities 25.1389(b) Fire-detection system 25.1203 (C) Pressure relief 25.1103(d) Emergency-locator transmitter (ELT) 25.1415(d)	47–400 onlý). (3) (747–400 only). See Note 1).) (See Note 1).

Requirement	Section
Emergency-exit marking	25.811(f).

If the Administrator finds that the applicable airworthiness regulations (i.e., 14 CFR part 25) do not contain adequate or appropriate safety standards for the 747–468 because of a novel or unusual design feature, special conditions are prescribed under the provisions of § 21.16.

In addition to the applicable airworthiness regulations and special conditions, the 747–468 must comply with the fuel-vent and exhaust-emission requirements of 14 CFR part 34, and the noise certification requirements of 14 CFR part 36.

The FAA issues special conditions, as defined in 14 CFR 11.19, in accordance with § 11.38, and they become part of the type-certification basis under § 21.101.

Special conditions are initially applicable to the model for which they are issued. Should the applicant apply for a supplemental type certificate to modify any other model included on the same type certificate to incorporate the same or similar novel or unusual design feature, the special conditions would also apply to the other model under § 21.101.

Novel or Unusual Design Features

The original aircraft configuration included a straight stairway between the main deck and upper deck at FS 870. The stairway is relocated in the new configuration, and the existing stairway is replaced with an electrically powered medical lift using the opening in the upper deck formerly occupied by the stairs. When the lift is not in operation, the upper-deck opening is covered by floor panels. These floor panels are opened up prior to operation of the lift and form a protective fencing around the upper-deck opening.

The purpose of the medical lift is to move an occupant between the master lounge in the upper deck and the medical room on the lower deck.

The lift platform is driven by two redundant electrical motors, mounted to the rear wall, between the struts. A lifting gear-drive with shafts and gear boxes is powered on the front and rear of the lift platform. The spindles are supported at the lifting gear on the lower support structure and with a strut support on the upper deck. The lift platform is guided in lateral directions with the guiding rails mounted on the struts.

Discussion

Due to the novel or unusual features associated with the installation of this medical lift, the following special conditions are considered necessary to provide a level of safety equal to that established by the airworthiness regulations incorporated by reference in the type-certificate.

Applicability

As discussed above, these special conditions are applicable to the 747– 468. Should JAES apply at a later date for a supplemental type certificate to modify any other model included on Type Certificate A20WE, to incorporate the same novel or unusual design feature, the special conditions would apply to that model as well.

Conclusion

This action affects only certain novel or unusual design features on one model of airplanes. It is not a rule of general applicability and it affects only the applicant who applied to the FAA for approval of these features on the airplane.

List of Subjects in 14 CFR Part 25

Aircraft, Aviation safety, Reporting and recordkeeping requirements.

The authority citation for these special conditions is as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701, 44702, 44704.

The Proposed Special Conditions

Accordingly, the Federal Aviation Administration (FAA) proposes the following special conditions as part of the type-certification basis for Boeing 747–468 airplanes modified by JAES.

1. A functional verification must be conducted to ensure the adequacy of the lift design features that are supposed to prevent injury to the lift occupant, lift operator, and lift observer.

2. The occupied lift must be designed to withstand the non-emergency load conditions imposed by the aircraft according to loads report SIE–327–301, revision D.

3. Occupancy or operation of the lift must not be permitted during taxi, takeoff, landing (TTL), or turbulent conditions.

4. The lift must be stowed for TTL. The stowed position requires the lift platform positioned at the main-deck level with the floor panels closed. 5. A portable oxygen bottle must be present in the lift and easily accessible to the occupant.

6. Occupancy of the lift must be limited to a single occupant secured in one of two possible configurations:

a. The occupant must be secured to a medical stretcher that is attached to the lift platform. The occupied stretcher must be designed to withstand the nonemergency load conditions defined in loads report SIE-327-301, revision D.

b. The occupant must be secured to a wheelchair that is attached to the lift platform.

7. Control panels must be located on both main and upper decks, connected with full duplex audio communications. On both operator control units, an emergency shut-off switch must be installed. In an emergency, this switch must immediately interrupt the main power supply to the motors. Lift operation must be stopped until the emergency shut-off switch is reset. As soon as one of the operators commands operation in a direction, the "Up" and "Down" option buttons must be disabled and the stop button enabled. Before one of the operators is able to change the lift-travel direction again, the lift must first be stopped.

8. Lift operation must require a trained operator at the main-deck control panel and a trained observer at the upper-deck control panel.

9. Sensors must be installed to detect the following conditions, and to prevent the start or continuation of lift travel if any conditions are not met:

a. Upper-deck seat, located on the left side of the aircraft and just forward of the master-bath bulkhead, is in its most forward, outboard position.

b. Upper-deck master-bedroom/ lavatory port bulkhead is opened and secured.

c. Upper-deck shower door is closed and secured.

d. Upper-deck master-lavatory door is opened and secured.

e. Upper-deck floor panels are opened and configured to form the protective fencing.

f. Main-deck inboard doors are closed and secured. The doors must be lockable only from the outside of the lift. This ensures that the operator has control of this area and that nobody is located under the lift.

g. Aircraft seat-belt-fasten signs must not be illuminated.

10. Sensors must be installed to detect the following conditions during

operation, and to prevent continued lift travel if any of these conditions occur:

a. Over-temperature of lift motors and/or power-frequency converter.

b. Presence of smoke at motors and in electrical-control cabinet.

c. Over-current at the lift motors. d. Asynchronous operation of the spindles.

¹11. A built-in fire extinguisher must be installed in the motor and electricalcontrol cabinet. This fire extinguisher must be designed to discharge automatically upon the occurrence of a fire.

12. The lift must have the provision for manual operation in the event of a malfunction such as a loss of power to the lift and/or associated systems.

13. A separate battery backup system must provide lighting for the lift-control system, lift control/sensors, communication system, and lift lights for a minimum of 10 minutes in the event of loss of power to the lift and/or associated systems.

14. Lift placards must be installed near or adjacent the control panels identified in special condition 7. The placards must be stated as follows:

a. THIS LIFT IS APPROVED FOR MOVING ONLY A SINGLE OCCUPANT BETWEEN THE MAIN AND UPPER DECKS AND ONLY WHEN SECURED TO EITHER AN APPROVED MEDICAL STRETCHER OR WHEELCHAIR. NO OTHER USES OF THIS LIFT ARE APPROVED.

b. DO NOT OPERATE LIFT DURING TAXI, TAKEOFF, LANDING, OR TURBULENCE.

c. AN APPROVED MEDICAL STRETCHER OR WHEELCHAIR MUST BE PROPERLY SECURED TO THE LIFT PLATFORM BEFORE OPERATING THIS LIFT.

d. THE LIFT MUST BE STOWED FOR TAXI, TAKEOFF, AND LANDING. THE STOWED POSITION REQUIRES THE LIFT PLATFORM POSITIONED AT THE MAIN–DECK LEVEL WITH THE FLOOR PANELS CLOSED.

15. Instructions on how to:

a. Configure the lift for operation.

b. Operate the lift.

c. Stow the lift for non-operation such as during TTL and turbulence.

d. Operate the mechanical-override features in the event of a malfunction such as a loss of power to the lift and/ or associated systems.

16. Training and related manuals must include:

a. Limitations and procedures for normal lift operation.

b. Backup and override procedure for evacuating the lift and returning it to TTL configuration. 17. Special conditions nos. 3. 4, and 14 must be documented in the Limitations section of the AFM.

Issued in Renton, Washington, on May 12, 2010.

Ali Bahrami,

Manager, Transport Airplane Directorate, Aircraft Certification Service. [FR Doc. 2010–11828 Filed 5–17–10; 8:45 am]

BILLING CODE 4910-13-P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA-2010-0515; Directorate Identifier 2009-NM-196-AD]

RIN 2120-AA64

Airworthiness Directives; Bombardier, Inc. Model CL–600–2C10 (Regional Jet Series 700, 701 & 702), Model CL–600– 2D15 (Regional Jet Series 705), and Model CL–600–2D24 (Regional Jet Series 900) Airplanes

AGENCY: Federal Aviation Administration (FAA), DOT. **ACTION:** Notice of proposed rulemaking (NPRM).

SUMMARY: We propose to adopt a new airworthiness directive (AD) for the products listed above. This proposed AD results from mandatory continuing airworthiness information (MCAI) originated by an aviation authority of another country to identify and correct an unsafe condition on an aviation product. The MCAI describes the unsafe condition as:

Several cases have been reported of cracks in the joint extrusions securing the outer bondment to the acoustic panel of the nacelle transcowl assemblies. Although there is no effect on flight safety (thrust reverser stowed), thrust reverser deployment under rejected take-off or emergency landing load conditions could potentially result in acoustic panel failure and possible runway debris.

* * * * *

The loss of an acoustic panel during rejected take-off or emergency landing load conditions could leave debris on the runway. This debris, if not removed, creates an unsafe condition for other airplanes during take-off or landing, as those airplanes could impact debris on the runway and sustain damage. The proposed AD would require actions that are intended to address the unsafe condition described in the MCAI.

DATES: We must receive comments on this proposed AD by July 2, 2010.

ADDRESSES: You may send comments by any of the following methods:

• Federal eRulemaking Portal: Go to http://www.regulations.gov. Follow the instructions for submitting comments.

• Fax: (202) 493-2251.

• *Mail:* U.S. Department of Transportation, Docket Operations, M–30, West Building Ground Floor, Room W12–140, 1200 New Jersey Avenue, SE., Washington, DC 20590.

• *Hand Delivery:* U.S. Department of Transportation, Docket Operations, M– 30, West Building Ground Floor, Room W12–40, 1200 New Jersey Avenue, SE., Washington, DC, between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

For service information identified in this proposed AD, contact Bombardier, Inc., 400 Côte-Vertu Road West, Dorval, Québec H4S 1Y9, Canada; telephone 514–855–5000; fax 514–855–7401; e-mail *thd.crj@aero.bombardier.com*; Internet *http://www.bombardier.com*. You may review copies of the referenced service information at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington. For information on the availability of this material at the FAA, call 425–227–1221.

Examining the AD Docket

You may examine the AD docket on the Internet at *http:// www.regulations.gov;* or in person at the Docket Operations office between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this proposed AD, the regulatory evaluation, any comments received, and other information. The street address for the Docket Operations office (telephone (800) 647–5527) is in the **ADDRESSES** section. Comments will be available in the AD docket shortly after receipt.

FOR FURTHER INFORMATION CONTACT:

Craig Yates, Aerospace Engineer, Airframe and Mechanical Systems Branch, ANE–171, FAA, New York Aircraft Certification Office, 1600 Stewart Avenue, Suite 410, Westbury, New York 11590; telephone (516) 228– 7355; fax (516) 794–5531.

SUPPLEMENTARY INFORMATION:

Comments Invited

We invite you to send any written relevant data, views, or arguments about this proposed AD. Send your comments to an address listed under the **ADDRESSES** section. Include "Docket No. FAA–2010–0515; Directorate Identifier 2009–NM–196–AD" at the beginning of your comments. We specifically invite comments on the overall regulatory,