Deadline for Filing Comments—March 14, 2011.

FOR FURTHER INFORMATION CONTACT: Teresa Hutchinson, Marketing Specialist, or Gary D. Olson, Regional Manager, Northwest Marketing Field Office, Marketing Order Administration Branch, Fruit and Vegetable Programs, AMS, USDA; Telephone: (503) 326–2724; Fax: (503) 326–7440, or E-mail: Teresa.Hutchinson@ams.usda.gov or Gary.D.Olson@ams.usda.gov, respectively.

SUPPLEMENTARY INFORMATION: Pursuant to Marketing Order No. 927 (7 CFR part 927), hereinafter referred to as the “order,” and the applicable provisions of the Agricultural Marketing Agreement Act of 1937, as amended (7 U.S.C. 601–674), hereinafter referred to as the “Act,” it is hereby directed that a referendum be conducted to ascertain whether continuance of the order is favored by growers. The referendum shall be conducted from February 26 through March 11, 2011, among eligible Oregon and Washington pear growers. Only growers that were engaged in the production of pears in Oregon or Washington during the period of July 1, 2009, through June 30, 2010, may participate in the continuance referendum.

USDA has determined that continuance referenda are an effective means for determining whether growers favor the continuation of marketing order programs. USDA would consider termination of the order if less than two-thirds of the growers voting in the referendum and growers of less than two-thirds of the volume of Oregon and Washington pears represented in the referendum favor continuance of their program. In evaluating the merits of continuance versus termination, USDA will not exclusively consider the results of the continuance referendum. USDA will also consider all other relevant information regarding operation of the order and relative benefits and disadvantages to growers, handlers, and consumers to determine whether continuing the order would tend to effectuate the declared policy of the Act.

In accordance with the Paperwork Reduction Act of 1995 (44 U.S.C. chapter 35), the ballot materials used in the referendum herein ordered have been submitted to and approved by the Office of Management and Budget (OMB) and have been assigned OMB No. 0581–0189. It has been estimated that it will take an average of 20 minutes for each of the approximately 1700 Oregon-Washington pear growers to cast a ballot. Participation is voluntary. Ballots postmarked after March 11, 2011, will not be included in the vote tabulation.

Teresa Hutchinson and Gary D. Olson, of the Northwest Marketing Field Office, Fruit and Vegetable Programs, AMS, USDA, are hereby designated as the referendum agents of the Secretary of Agriculture to conduct this referendum. The procedure applicable to the referendum shall be the “Procedure for the Conduct of Referenda in Connection With Marketing Orders for Fruits, Vegetables, and Nuts Pursuant to the Agricultural Marketing Agreement Act of 1937, as Amended” (7 CFR §§ 900.400–900.407).

Ballots will be mailed to all growers of record and may also be obtained from the referendum agents or from their appointees.

List of Subjects in 7 CFR Part 927

Marketing agreements and orders, Pears, Reporting and recordkeeping.


Rayne Pegg, Administrator, Agricultural Marketing Service.

[FR Doc. 2011–3501 Filed 2–15–11; 8:45 am]
BILLING CODE 3410–02–P
ADDRESSSES: You must mail two copies of your comments to: Federal Aviation Administration, Transport Airplane Directorate, Attn: Rules Docket (ANM–113), Docket No. NM445, 1601 Lind Avenue, SW., Renton, Washington 98057–3356. You may deliver two copies to the Transport Airplane Directorate at the above address. You must mark your comments: Docket No. NM445. You can inspect comments in the Rules Docket weekdays, except Federal holidays, between 7:30 a.m. and 4 p.m.


SUPPLEMENTARY INFORMATION:

Comments Invited

We invite interested people to take part in this rulemaking by sending written comments, data, or views. The most helpful comments reference a specific portion of the special conditions, explain the reason for any recommended change, and include supporting data. We ask that you send us two copies of written comments.

We will file in the docket all comments we receive, as well as a report summarizing each substantive public contact with FAA personnel concerning these special conditions. You can inspect the docket before and after the comment closing date. If you wish to review the docket in person, go to the address in the ADDRESSSES section of this preamble between 7:30 a.m. and 4 p.m., Monday through Friday, except Federal holidays.

We will consider all comments we receive on or before the closing date for comments. We will consider comments filed late if it is possible to do so without incurring expense or delay. We may change these special conditions based on the comments we receive. If you want us to acknowledge receipt of your comments on this proposal, include with your comments a self-addressed, stamped postcard on which you have written the docket number. We will stamp the date on the postcard and mail it back to you.

Background

On March 29, 2005, Gulfstream Aerospace Corporation (hereafter referred to as “Gulfstream”) applied for an FAA type certificate for its new Gulfstream Model GVI passenger airplane. Gulfstream later applied for, and was granted, an extension of time for the type certificate, which changed the effective application date to September 28, 2006. The Gulfstream Model GVI airplane will be an all-new, two-engine jet transport airplane with an executive cabin interior. The maximum takeoff weight will be 99,600 pounds, with a maximum passenger count of 19 passengers.

Type Certification Basis

Under provisions of Title 14, Code of Federal Regulations (14 CFR) 21.17, Gulfstream must show that the Gulfstream Model GVI airplane (hereafter referred to as “the GVI”) meets the applicable provisions of 14 CFR part 25, as amended by Amendments 25–1 through 25–119, 25–122, and 25–124. If the Administrator finds that the applicable airworthiness regulations (i.e., 14 CFR part 25) do not contain adequate or appropriate safety standards for the GVI because of a novel or unusual design feature, special conditions are prescribed under the provisions of § 21.16.

In addition to complying with the applicable airworthiness regulations and special conditions, the GVI must comply with the fuel vent and exhaust emission requirements of 14 CFR part 34 and the noise certification requirements of 14 CFR part 36. The FAA must also issue a finding of regulatory adequacy pursuant to section 611 of Public Law 92–574, the “Noise Control Act of 1972.” The FAA issues special conditions, as defined in 14 CFR 11.19, in accordance with § 11.38, and they become part of the type certification basis under § 21.17(a)(2).

Special conditions are initially applicable to the model for which they are issued. Should the type certificate for that model be amended later to include another model that incorporates the same novel or unusual design features, the special conditions would also apply to the other model under provisions of §21.101.

Novel or Unusual Design Features

The GVI is equipped with a high speed protection system that limits nose down pilot authority at speeds above Vc/Mc, and prevents the airplane from actually performing the maneuver required under § 25.335(b)(1). The applicable airworthiness regulations do not contain adequate or appropriate safety standards for this design feature. These proposed special conditions address this design feature. These proposed special conditions are identical or nearly identical to those previously required for type certification of other airplane models.

Discussion of Proposed Special Conditions

Gulfstream proposes to reduce the speed margin between Vc and Vp, required by § 25.335(b), based on the incorporation of a high speed protection system in the GVI flight control laws. The GVI is equipped with a high speed protection system that limits nose down pilot authority at speeds above Vc/Mc, and prevents the airplane from actually performing the maneuver required under § 25.335(b)(1).

Section 25.335(b)(1) is an analytical envelope condition which was originally adopted in Part 4b of the civil air regulations to provide an acceptable speed margin between design cruise speed and design dive speed. Freedom from flutter and the airframe design loads are affected by the design dive speed. While the initial condition for the upset specified in the rule is 1g level flight, protection is afforded for other inadvertent overspeed conditions as well. Section 25.335(b)(1) is intended as a conservative enveloping condition for all potential overspeed conditions, including non-symmetric ones.

To establish that all potential overspeed conditions are enveloped, the applicant would demonstrate that the dive speed will not be exceeded during pilot-induced or gust-induced upsets in non-symmetric attitudes.

In addition, the high speed protection system in the GVI must have a high level of reliability.

Applicability

As discussed above, these proposed special conditions are applicable to the GVI. Should Gulfstream apply at a later date for a change to the type certificate to include another model incorporating the same novel or unusual design features, these proposed special conditions would apply to that model as well.

Conclusion

This action affects only certain novel or unusual design features of the GVI. It is not a rule of general applicability.

List of Subjects in 14 CFR Part 25

Aircraft, Aviation safety, Reporting and recordkeeping requirements.

The authority citation for these special conditions is as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701, 44702, 44704.

The Proposed Special Conditions

Accordingly, the Federal Aviation Administration (FAA) proposes the
following special conditions as part of the type certification basis for the GVI airplanes.

1. In lieu of compliance with § 25.335(b)(1), if the flight control system includes functions that act automatically to initiate recovery before the end of the 20 second period specified in § 25.335(b)(1), $V_{CM}/M_C$ must be determined from the greater of the speeds resulting from conditions (a) and (b) below. The speed increase occurring in these maneuvers may be calculated if reliable or conservative aerodynamic data are used.

   (a) From an initial condition of stabilized flight at $V_c/M_c$, the airplane is upset so as to take up a new flight path 7.5 degrees below the initial path. Control application, up to full authority, is made to try to maintain this new flight path. Twenty seconds after initiating the upset, manual recovery is initiated, at which time power reduction and the use of pilot controlled drag devices may be used.

   (b) From a speed below $V_c/M_c$, with power to maintain stabilized level flight at this speed, the airplane is upset so as to accelerate through $V_c/M_c$ at a flight path 15 degrees below the initial path (or at the steepest nose down attitude that the system will permit with full control authority if less than 15 degrees). The pilot’s controls may be in the neutral position after reaching $V_c/M_c$ and before recovery is initiated. Recovery may be initiated three seconds after operation of high speed warning system by application of a load factor of 1.5g (0.5 acceleration increment), or such greater load factor that is automatically applied by the system with the pilot’s pitch control neutral. Power, as specified in § 25.175(b)(1)(iv), is assumed until recovery is initiated, at which time power reduction and the use of pilot controlled bomb racks may be used.

2. The applicant must also demonstrate that the speed margin, established as above, will not be exceeded in inadvertent or gust induced upsets resulting in initiation of the dive from non-symmetric attitudes, unless the airplane is protected by the flight control laws from getting into non-symmetric conditions. The upset maneuvers described in Advisory Circular 25–7A, Change 1, section 32, paragraphs c.(1)(i) and (iii) may be used to comply with this requirement.

3. Any failure of the high speed protection system that would affect the speed margin determined by paragraphs 1 and 2. must be improbable (occur at a rate less than $10^{-5}$ per flight hour).

4. Failures of the system must be annunciated to the pilots, and flight manual instructions must be provided to reduce the maximum operating speeds, $V_{MO}/M_{MO}$. The operating speed must be reduced to a value that maintains a speed margin between $V_{MO}/M_{MO}$ and $V_{D}/M_{D}$ that is consistent with showing compliance with § 25.335(b) without the benefit of the high speed protection system.

5. Master minimum equipment list (MMEL) relief for the high speed protection system may be considered by the FAA Flight Operations Evaluation Board (FOEB) provided that the flight manual instructions indicate reduced maximum operating speeds as described in paragraph 4., and that no additional hazards are introduced with the high speed protection system inoperative. In addition, the cockpit display of the reduced operating speeds, as well as the overspeed warning for exceeding those speeds, must be equivalent to that of the normal airplane with the high speed protection system operative.

Issued in Renton, Washington, on February 3, 2011.

Jeffrey E. Duven,
Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. 2011–3412 Filed 2–15–11; 8:45 am]

BILLING CODE 4910–13–P

DEPARTMENT OF TRANSPORTATION
Federal Aviation Administration

14 CFR Part 39

RIN 2120–AA64

Airworthiness Directives; DASSAULT AVIATION Model MYSTERE-FALCON 50 Airplanes

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Notice of proposed rulemaking (NPRM).

SUMMARY: We propose to adopt a new airworthiness directive (AD) for the products listed above that would supersede an existing AD. This proposed AD results from the mandatory continuing airworthiness information (MCAI) originated by an aviation authority of another country to identify and correct an unsafe condition on an aviation product. The MCAI describes the unsafe condition as:

On two occurrences on Mystère-Falcon 50 aeroplanes in service, it was detected that two pipes of the emergency brake system #2 located near the nose landing gear bearing were swapped.

The swapping of these two pipes implies that when the Left Hand (LH) brake pedal is depressed, the Right Hand (RH) brake unit is activated, and conversely, when the RH brake pedal is depressed, the LH brake unit is activated. This constitutes an unsafe condition, which may go unnoticed as the condition is latent until the emergency brake system #2 is used. This condition, if not corrected, could ultimately lead to a runway excursion of the aeroplane.

* * * * *

The proposed AD would require actions that are intended to address the unsafe condition described in the MCAI.

DATES: We must receive comments on this proposed AD by April 4, 2011.

ADDRESSES: You may send comments by any of the following methods:

• Federal eRulemaking Portal: Go to http://www.regulations.gov. Follow the instructions for submitting comments.

• Fax: (202) 493–2251.

• Mail: U.S. Department of Transportation, Docket Operations, M–30, West Building Ground Floor, Room W12–140, 1200 New Jersey Avenue, SE., Washington, DC 20590.

• Hand Delivery: U.S. Department of Transportation, Docket Operations, M–30, West Building Ground Floor, Room W12–140, 1200 New Jersey Avenue, SE., Washington, DC, between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

For service information identified in this proposed AD, contact Dassault Falcon Jet, P.O. Box 2000, South Hackensack, New Jersey 07606; telephone 201–440–6700; Internet http://www.dassaultfalcon.com. You may review copies of the referenced service information at the FAA, Transport Airplane Directorate, 1601 Lind Avenue SW., Renton, Washington. For information on the availability of this material at the FAA, call 425–227–1221.

Examining the AD Docket

You may examine the AD docket on the Internet at http://www.regulations.gov; or in person at the Docket Operations office between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this proposed AD, the regulatory evaluation, any comments received, and other information. The street address for the Docket Operations