
Small Business Regulatory Enforcement Fairness Act

The Small Business Regulatory Enforcement Fairness Act of 1996 (Pub. L. 104–121) provides generally for congressional review of agency rules. A reporting requirement is triggered in instances where NCUA issues a final rule as defined by Section 551 of the APA. 5 U.S.C. 551. The Office of Management and Budget has determined that this rule is not a major rule for purposes of the Small Business Regulatory Enforcement Fairness Act of 1996.

Dated: By the National Credit Union Administration Board on May 24, 2012.

Mary F. Rupp,
Secretary of the Board.

Accordingly, for the reasons set forth in the preamble, IRPS 12–1 amends IRPS 11–1 as follows:

Note: The following ruling will not appear in the Code of Federal Regulations.


2. Amend the third paragraph in subpart A to read as follows:

Material supervisory determinations are limited to: (1) Composite CAMEL ratings of 3, 4, and 5 and all component ratings of those composite ratings; (2) adequacy of loan loss reserve provisions; and (3) loan classifications on loans that are significant as determined by the appealing credit union. Subject to the requirements discussed below, credit unions may also appeal to the Committee a decision of the Director of the Office of Small Credit Union Initiatives (OSCUI) to deny Technical Assistance Grant (TAG) reimbursements.

3. Remove the 7th paragraph in subpart A.

4. Revise the last paragraph in subpart A to read as follows:

Committee decisions on the denial of a TAG reimbursement are the final decisions of NCUA and are not appealable to the NCUA Board. All other appealable decisions must be appealed to the NCUA Board within 30 days of the appellant’s receipt by the party of the Committee’s decision.

[FR Doc. 2012–13210 Filed 5–30–12; 8:45 am]

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 25

[Docket No. NM438 Special Conditions No. 25–423–SC]

Special Conditions: Gulfstream Model GVI Airplane; High Incidence Protection

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Final special conditions; correction.

SUMMARY: This document corrects an error that appeared in Docket No. NM438, Special Conditions No. 25–423–SC, which were published in the Federal Register on March 28, 2011. The error resulted in the omission of two paragraphs of text in The Special Conditions section.


SUPPLEMENTARY INFORMATION: The document designated as “Docket No. NM438, Special Conditions No. 25–423–SC” was published in the Federal Register on March 28, 2011 (76 FR 17022). The document issued special conditions pertaining to a high incidence protection system that replaces the stall warning system during normal operating conditions, prohibits the airplane from stalling, limits the angle of attack at which the airplane can be flown during normal low speed operations, and cannot be overridden by the flight crew. These special conditions were, and continue to be applicable to, Gulfstream Model GVI airplanes.

As published, the document contained an error because paragraphs 3(e)(6) and 3(e)(7) were omitted. Due to its complexity the entire text of paragraph 3(e) is included below, including paragraphs 3(e)(6) and 3(e)(7).

3. Minimum Steady Flight Speed and Reference Stall Speed—In lieu of the requirements of § 25.103, the following special condition is issued:

(e) $V_{SR}$ must be determined with the following conditions:

(1) Engines idling, or, if that resultant thrust causes an appreciable decrease in stall speed, not more than zero thrust at the stall speed.

(2) The airplane in other respects (such as flaps and landing gear) in the condition existing in the test or performance standard in which $V_{SR}$ is being used.

(3) The weight used when $V_{SR}$ is being used as a factor to determine compliance with a required performance standard.

(4) The center of gravity position that results in the highest value of reference stall speed.

(5) The airplane trimmed for straight flight at a speed selected by the applicant, but not less than 1.13 $V_{SR}$ and not greater than 1.3 $V_{SR}$.

(6) The high incidence protection function disabled, or adjusted to a high enough incidence to allow full development of the maneuver to the angle of attack corresponding to $V_{SR}$.

(7) From the stabilized trim condition, apply the longitudinal control to decelerate the airplane so that the speed reduction does not exceed one knot per second.

Since no other part of the regulatory information has been changed, the special conditions are not being republished.

Correction

In Final special conditions document [FR Doc. 2011–7144 Filed 3–25–11; 8:45 a.m.] published on March 28, 2011 (76 FR 17022), make the following correction:

On page 17024, in the first column, which begins with (e), include the following paragraphs after (5) and before (f):

6. The high incidence protection function disabled, or adjusted to a high enough incidence to allow full development of the maneuver to the angle of attack corresponding to $V_{SR}$.

(7) From the stabilized trim condition, apply the longitudinal control to decelerate the airplane so that the speed reduction does not exceed one knot per second.

Issued in Renton, Washington, on May 18, 2012.

Michael J. Kaszynski,
Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.

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