(f) Special Flight Permits

Special flight permits will not be issued.

(g) Alternative Methods of Compliance (AMOCs)

(1) The Manager, Safety Management Group, FAA, may approve AMOCs for this AD. Send your proposal to: Robert Grant, Aviation Safety Engineer, Safety Management Group, FAA, 2601 Meacham Blvd., Fort Worth, Texas 76137; telephone 817–222–5326; email robert.grant@faa.gov.

(2) For operations conducted under a 14 CFR part 119 operating certificate or under 14 CFR part 91, subpart K, we suggest that you notify your principal inspector, or lacking a principal inspector, the manager of the local flight standards district office or certificate holding district office, before operating any aircraft complying with this AD through an AMOC.

(h) Additional Information

(1) Agusta Bollettino Tecnico No. 139–308, dated October 16, 2012, which is not incorporated by reference, contains additional information about the subject of this AD. For a copy of the service information referenced in this AD, contact:

Giovanni Cecchelli; telephone 39–0331–711133; fax 39 0331 711180; or at http://www.agustawestland.com/technical-bulletins. You may review a copy of the service information at the FAA, Office of the Regional Counsel, Southwest Region, 2601 Meacham Blvd., Room 663, Fort Worth, Texas 76137.

(2) The subject of this AD is discussed in European Aviation Safety Agency AD No. 2012–0213–E, dated October 16, 2012.

(i) Subject


Issued in Fort Worth, Texas, on December 3, 2012.

Kim Smith,

Directorate Manager, Rotorcraft Directorate, Aircraft Certification Service.

[FR Doc. 2012–29716 Filed 12–7–12; 8:45 am]
warning in icing conditions, and modifying the airplane for the replacement of the SWC. We are issuing this AD to prevent natural stall events when operating in icing conditions, which, if not corrected, could result in loss of control of the airplane.

DATES: This AD becomes effective January 14, 2013.

The Director of the Federal Register approved the incorporation by reference of certain publications listed in this AD as of January 14, 2013.

ADDRESSES: You may examine the AD docket on the Internet at http://www.regulations.gov or in person at the U.S. Department of Transportation, Docket Operations, M–30, West Building Ground Floor, Room W12–140, 1200 New Jersey Avenue SE., Washington, DC.

FOR FURTHER INFORMATION CONTACT:

SUPPLEMENTAL INFORMATION:

Discussion

We issued a notice of proposed rulemaking (NPRM) to amend 14 CFR part 39 to include an AD that would apply to the specified products. That NPRM was published in the Federal Register on June 27, 2012 (77 FR 38224). That NPRM proposed to correct an unsafe condition for the specified products. The Mandatory Continuing Airworthiness Information (MCAI) states:

A few natural stall events, specifically when operating in icing conditions, have been experienced on SAAB 340 series aeroplanes, without receiving a prior stall warning. This condition, if not corrected, could result in loss of control of the aeroplane.

To address this potential unsafe condition, a modified stall warning system, incorporating improved stall warning logic, has been developed.

SAAB have issued Service Bulletin (SB) 340–27–098 and SB 340–27–099, which include instructions to replace the present Stall Warning Computer (SWC) with a new SWC, and instructions to activate the new SWC. The new system includes stall warning curves optimized for operation in icing conditions, which are activated by selection of Engine Anti-Ice.

For the reasons described above, this [European Aviation Safety Agency (EASA)] AD requires the replacement of the SWC, by installing new SWC Part Number (P/N) 0020AK6 on aeroplanes with basic wing tip, and installing a new SWC P/N 0020AK7 on aeroplanes with extended wing tip, as applicable to aeroplane configuration.

You may obtain further information by examining the MCAI in the AD docket.

Comments

We gave the public the opportunity to participate in developing this AD. We considered the comments received. The National Transportation Safety Board stated that it supports the NPRM (77 FR 38224, June 27, 2012).

Request To Revise Company Name and Email Address

Saab AB (the commenter) requested that we revise the NPRM (77 FR 38224, June 27, 2012) throughout to change the name of the Saab company referenced in the NPRM from “Saab AB, Saab Aerosystems” to the new company name, “Saab AB, Aeronautics.” The commenter also requested to change the company email address referenced in the NPRM from “saab2000itech support@saabgroup.com” to “saab340 techsupport@saabgroup.com.”

We agree because there is a potential problem with availability of the new name, “Saab AB, Aeronautics” to the new company name, “Saab AB, Aeronautics.” The commenter also requested that the email address be changed appropriately in subsequent ADs. We changed the email address appropriately in paragraph (m) of this AD.

Request To Revise Text in Summary Section and Paragraph (e) of the NPRM (77 FR 38224, June 27, 2012)

Saab AB requested that we revise the second and third sentences in the Summary section and a related sentence in paragraph (e) of the NPRM (77 FR 38224, June 27, 2012). In the Summary section, the commenter requested that we revise the following text to clarify that the airplane had been certified with natural buffet—as the stall warning—in icing conditions, and explained that replacement of the SWC would increase the safety level by introducing an artificial stall warning for icing conditions.

This proposed AD was prompted by reports of stall events during icing conditions which were not accompanied with a prior stall warning. This proposed AD would require replacing the stall warning computer (SWC) with a new SWC, and modifying the airplane for the replacement of the SWC.

The commenter suggested the following text.

This proposed AD was prompted by reports of stall events during icing conditions where the natural stall warning (buffet) was not identified. This proposed AD would require replacing the stall warning computer (SWC) with a new SWC, which provides an artificial stall warning in icing conditions, and modifying the airplane for the replacement of the SWC.

The commenter also requested that the first sentence in paragraph (e) of the NPRM (77 FR 38224, June 27, 2012) be changed to match the new wording.

We agree with the request for the reasons provided by the commenter. Therefore, we changed the wording in the Summary section and paragraph (e) of this AD as proposed by the commenter.

Request To Revise Text in Paragraph (i) of the NPRM (77 FR 38224, June 27, 2012)

Saab AB requested that we revise paragraph (i) of the NPRM (77 FR 38224, June 27, 2012), which reads “As of effective date of this AD, do not install any SWC having P/N 0020AK, 0020AK1, 0020AK2, 0020AK4, or 0020AK3 MOD 1 on any airplane,” to “After modification of an airplane as required by paragraph (g) and (h) of this AD, do not install any SWC having P/N 0020AK, 0020AK1, 0020AK2, 0020AK4, or 0020AK3 MOD 1 on any airplane.” The commenter stated that this change would allow an operator, during the compliance time, to install an SWC having the same part number of the failed SWC.

We agree because there is a potential problem with availability of the new part. Therefore, we have changed the wording in paragraph (i) of this AD to match the intent of the commenter’s request.

Conclusion

We reviewed the available data, including the comments received, and determined that air safety and the public interest require adopting the AD with the changes described previously—except for minor editorial changes. We have determined that these changes:

• Are consistent with the intent that was proposed in the NPRM (77 FR 38224, June 27, 2012) for correcting the unsafe condition; and
• Do not add any additional burden upon the public than was already proposed in the NPRM (77 FR 38224, June 27, 2012).

Costs of Compliance

We estimate that this AD will affect 162 products of U.S. registry. We also estimate that it will take about 78 work-hours per product to comply with the basic requirements of this AD. The average labor rate is $85 per work-hour. Required parts will cost about $33,000 per product. Where the service information lists required parts costs that are covered under warranty, we
have assumed that there will be no charge for these parts. As we do not control warranty coverage for affected parties, some parties may incur costs higher than estimated here. Based on these figures, we estimate the cost of this AD to the U.S. operators to be $6,420,060, or $39,630 per product.

Authority For This Rulemaking

Title 49 of the United States Code specifies the FAA’s authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. “Subtitle VII: Aviation Programs,” describes in more detail the scope of the Agency’s authority.

We are issuing this rulemaking under the authority described in “Subtitle VII, Part A, Subpart III, Section 44701: General requirements.” Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

Regulatory Findings

We determined that this AD will not have federalism implications under Executive Order 13132. This AD will not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify that this AD:

1. Is not a “significant regulatory action” under Executive Order 12866;
2. Is not a “significant rule” under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979);
3. Will not affect intrastate aviation in Alaska; and
4. Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

We prepared a regulatory evaluation of the estimated costs to comply with this AD and placed it in the AD docket.

Examining the AD Docket

You may examine the AD docket on the Internet at http://www.regulations.gov; or in person at the Docket Operations office between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains the NPRM (77 FR 38224, June 27, 2012), the regulatory evaluation, any comments received, and other information. The street address for the Docket Operations office (telephone (800) 647–5527) is in the ADDRESSES section. Comments will be available in the AD docket shortly after receipt.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

Adoption of the Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA amends 14 CFR part 39 as follows:

PART 39—AIRWORTHINESS DIRECTIVES

§ 39.13 [Amended]

2. The FAA amends § 39.13 by adding the following new AD:


(a) Effective Date

This airworthiness directive (AD) becomes effective January 14, 2013.

(b) Affected ADs

None.

(c) Applicability

This AD applies to Saab AB, Saab Aerosystems Model 340A (SAAB/340A) and SAAB 340B airplanes, certified in any category, as identified in paragraphs (c)(1) and (c)(2) of this AD, except airplanes that have SAAB Modification 2650 and/or 2859 installed. This AD does not apply to airplanes with serial numbers 170, 342, 362, 363, 367, 372, 379, 385, 395, 405, 409, 431, and 455.

(1) Model 340A (SAAB/340A) airplanes, serial numbers 004 through 159 inclusive.

(2) Model SAAB 340B airplanes, serial numbers 160 through 459 inclusive.

(d) Subject

Air Transport Association (ATA) of America Code 27: Flight Controls.

(e) Reason

This AD was prompted by reports of stall events during icing conditions where the natural stall warning (buffet) was not observed. We are issuing this AD to prevent natural stall events when operating in icing conditions, which, if not corrected, could result in loss of control of the airplane.

(f) Compliance

You are responsible for having the actions required by this AD performed within the compliance times specified, unless the actions have already been done.

Replace (g) Replacement

(1) For airplanes with basic wing tips:

Within 24 months after the effective date of this AD, replace all stall warning computers (SWCs) having part number (P/N) 0020AK, 0020AK1, 0020AK2, or 0020AK4, with a new SWC P/N 0020AK6, in accordance with the Accomplishment Instructions of Saab Service Bulletin 340–27–097, Revision 01, dated April 13, 2012.

(2) For airplanes with extended wing tips:

Within 24 months after the effective date of this AD, replace the SWC P/N 0020AK3 MOD 1 with a new SWC P/N 0020AK7, in accordance with the Accomplishment Instructions of Saab Service Bulletin 340–27–099, Revision 01, dated April 13, 2012.

(b) Concurrent Modification

Before or concurrently with the accomplishment of the requirements of paragraph (g) of this AD: Modify the airplane in accordance with the Accomplishment Instructions of Saab Service Bulletin 340–27–097, Revision 03, dated April 19, 2012.

(i) Parts Installation Prohibition

After accomplishing the replacement required by paragraph (g) of this AD and the modification required by paragraph (h) of this AD, do not install any SWC having P/N 0020AK, 0020AK1, 0020AK2, 0020AK4, or 0020AK3 MOD 1 on any airplane.

(j) Credit for Previous Actions

This paragraph provides credit for actions required by paragraph (h) of this AD, if those actions were performed before the effective date of this AD using the service bulletin specified in paragraph (j)(1), (j)(2), or (j)(3) of this AD, which are not incorporated by reference in this AD.


(k) Other FAA AD Provisions

The following provisions also apply to this AD:

(1) Alternative Methods of Compliance (AMOCs): The Manager, ANM–116, International Branch, Transport Airplane Directorate, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or local Flight Standards District Office, as appropriate. If sending information directly to the International Branch, send it to ATTN: Shahram Daneshmandi, Aerospace Engineer, International Branch, ANM–116, Transport Airplane Directorate, FAA, 1601 Lincoln Avenue SW., Renton, Washington 98057–3356; telephone (425) 227–1112; fax (425) 227–1149. Information may be emailed to: 9-ANM–116-AMOC-REQUESTS@faa.gov.

Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the local flight standards district office/
DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39


RIN 2120–AA64

Airworthiness Directives; The Boeing Company Airplanes

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Final rule.

SUMMARY: We are adopting a new airworthiness directive (AD) for certain The Boeing Company Model 737–600, –700, –700C, –800, –900, and –900ER series airplanes. This AD was prompted by reports of flight crew failure to activate air data probe heat. This AD requires modifying the anti-icing system for the angle of attack sensor, the total air temperature, and the pitot probes. We are issuing this AD to prevent ice from forming on air data system sensors and consequent loss of or misleading airspeed indication on all airspeed and consequent loss of or misleading altimeter indication.

DATES: This AD is effective January 14, 2013.

Discussion

We issued a notice of proposed rulemaking (NPRM) to amend 14 CFR part 39 to include an AD that would apply to the specified products. That NPRM published in the Federal Register on February 28, 2012 (77 FR 11789). That NPRM proposed to require modifying the anti-icing system for the angle of attack sensor, the total air temperature, and the pitot probes.

Actions Since Issuance of NPRM (77 FR 11789, February 28, 2012)


Comments

We gave the public the opportunity to participate in developing this AD. The following presents the comments received on the proposal (77 FR 11789, February 28, 2012) and the FAA’s response to each comment. Aviation Partners Boeing stated that installation of winglets per STC ST00830SE does not affect the NPRM and stated it would provide supporting data to the FAA upon request.

Request To Revise Compliance Time

Seven commenters requested that we revise the 24-month compliance time, as proposed in the NPRM (77 FR 11789, February 28, 2012).

Korean Air Lines (KAL), Air Pacific Limited (APC), Delta Air Lines, Inc. (DAL), American Airlines (AAL), United Airlines (UAL), and Southwest Airlines (SWA) requested we extend the compliance time. UAL and AAL requested we take into account the time needed to obtain modification kits and...