SUMMARY: Programs’ Drug Testing Requirements
Revisions to Fitness for Duty
RIN 3150–AI67

[81x482]Nuclear Regulatory
Agency: Nuclear Regulatory
Commission. ACTION: Regulatory basis.

SUMMARY: The U.S. Nuclear Regulatory
Commission (NRC) is making available
the regulatory basis for the ongoing
proposed rulemaking effort to amend its
regulations regarding drug testing
requirements in NRC licensees’ fitness
for duty programs. The regulatory basis
documents the reasoning upon which
the NRC determined rulemaking was the
appropriate course of action. In this
regulatory basis, the NRC recommends
developing a proposed rule that would
enhance the ability of NRC licensees to
detect and deter drug use and the
alignment of the NRC’s regulations with
select drug testing provisions in the U.S.
Department of Health and Human
Services’ “Mandatory Guidelines for
Federal Workplace Drug Testing
Programs’” issued in 2008.

DATES: At this time, the NRC is not
soliciting formal public comments on the
materials identified in this
document. There will be an opportunity
for formal public comment on the
proposed rule when it is published in the
Federal Register.

ADRESSES: Please refer to Docket ID
NRC–2009–0225 when contacting the
NRC about the availability of
information regarding this document.
You may access information related to
this document, which the NRC
possesses and is publicly available,
using any of the following methods:

• Federal Rulemaking Web site: Go to
http://www.regulations.gov and search
for Docket ID NRC–2009–0225. Address
questions about NRC dockets to Carol
Gallagher; telephone: 301–492–3668;
email: Carol.Gallagher@nrc.gov. For
technical questions, contact the
individuals listed in the FOR FURTHER
INFORMATION CONTACT section of this
document.

• NRC’s Agencywide Documents
Access and Management System
(ADAMS): You may access publicly
available documents online in the NRC
Library at http://www.nrc.gov/reading-
rm/adams.html. To begin the search,
select “ADAMS Public Documents” and
then select “Begin Web-based ADAMS
Search.” For problems with ADAMS,
please contact the NRC’s Public
Document Room (PDR) reference staff at
1–800–397–4209, 301–415–4737, or by
e-mail to pdr.resource@nrc.gov. The
ADAMS accession number for each
document referenced in this document
(if that document is available in
ADAMS) is provided the first time that
a document is referenced.

• NRC’s PDR: You may examine and
purchase copies of public documents at
the NRC’s PDR, Room O1–F21, One
White Flint North, 11555 Rockville
Pike, Rockville, Maryland 20852.

FOR FURTHER INFORMATION CONTACT:
Scott C. Sloan, Office of Nuclear Reactor
Regulation, U.S. Nuclear Regulatory
Commission, Washington, DC 20555–
0001; telephone: 301–415–1619; email:
Scott.Sloan@nrc.gov.

SUPPLEMENTARY INFORMATION:

Additional Documents and Public
Meetings

As the NRC continues its ongoing
proposed rulemaking effort to amend
the drug testing requirements of part 26
of Title 10 of the Code of Federal
Regulations (10 CFR), the NRC will
periodically make preliminary draft
documents publicly available on the
Federal rulemaking Web site,
www.regulations.gov, under docket ID
NRC–2009–0225. The availability of
these documents informs stakeholders
of the current status of the NRC’s
rulemaking development activities and
provides preparatory material for future
public meetings. The NRC is not
instituting a public comment period on
these materials, but the public is
encouraged to participate in related
public meetings. In addition, the public
will be given ample opportunity to
provide comments on the proposed rule
upon its publication in the Federal
Register. The NRC will post meeting
notices to the NRC’s Public Meeting
public-involve/public-meetings/
index.cfm, 10 days prior to any meeting
dates. Additional documents related to
this proposed rulemaking, including
meeting notices, will be made publicly
available on the Federal rulemaking
Web site at https://www.regulations.gov,
under Docket ID NRC–2009–0225. The
Federal rulemaking Web site allows you
to receive alerts when changes or
additions occur in a docket folder. To
subscribe: (1) Navigate to the docket
folder (NRC–2009–0225); (2) click the
“Email Alert” link; and (3) enter your
email address and select how frequently
you would like to receive emails (daily,
weekly, or monthly).

Dated at Rockville, Maryland, this 19th
day of June, 2013.

For the Nuclear Regulatory Commission.

Lawrence E. Kokajko,
Director, Division of Policy and Rulemaking.

Federal Aviation Administration

DEPARTMENT OF TRANSPORTATION

Airworthiness Directives; Airbus
Airplanes

AGENCY: Federal Aviation
Administration (FAA), DOT.

ACTION: Notice of proposed
rulemaking (NPRM).

SUMMARY: We propose to adopt a new
airworthiness directive (AD) for all
Airbus Model A318, A319, A320, and
A321 series airplanes. This proposed
AD was prompted by a determination
that certain special washers used in the
retraction jack anchorage fitting bearing
installation in the main landing gear
(MLG) were incorrectly manufactured.
This proposed AD would require an
inspection of the left-hand (LH) and
right-hand (RH) MLG retraction jack
anchorage fitting bearing assemblies to
verify that the special washer is seated
correctly, and related investigative and
corrective actions if necessary. We are
proposing this AD to detect and correct

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[81x444]Federal Aviation
Administration

Federal Register

Vol. 78, No. 126

Monday, July 1, 2013

Airworthiness Directives; Airbus
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anchorage fitting bearing assemblies to
verify that the special washer is seated
correctly, and related investigative and
corrective actions if necessary. We are
proposing this AD to detect and correct
installation of incorrectly manufactured special washers, which could lead to a local stress concentration resulting in possible reduction of the fatigue life of the jack fitting, and consequent reduction of the structural integrity of the affected MLG.

DATES: We must receive comments on this proposed AD by August 15, 2013.

ADDRESSES: You may send comments by any of the following methods:
• Federal eRulemaking Portal: Go to http://www.regulations.gov. Follow the instructions for submitting comments.
• Fax: (202) 493–2251.
• Hand Delivery: U.S. Department of Transportation, Docket Operations, M–30, West Building Ground Floor, Room W12–140, 1200 New Jersey Avenue SE., Washington, DC, between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

For service information identified in this proposed AD, contact Airbus, Airworthiness Office—EIAF, 1 Rond Point Maurice Bellonte, 31707 Blagnac Cedex, France; telephone +33 5 61 93 36 96; fax +33 5 61 93 44 51; email account.airworth-eas@airbus.com; Internet http://www.airbus.com. You may review copies of the referenced service information at the FAA, Transport Airplane Directorate, 1601 Lind Avenue SW., Renton, WA. For information on the availability of this material at the FAA, call 425–227–1221.

Examining the AD Docket
You may examine the AD docket on the Internet at http://www.regulations.gov; or in person at the Docket Operations office between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this proposed AD, the regulatory evaluation, any comments received, and other information. The street address for the Docket Operations office (telephone (800) 647–5527) is in the ADDRESSES section. Comments will be available in the AD docket shortly after receipt.

FOR FURTHER INFORMATION CONTACT:

SUPPLEMENTARY INFORMATION:
Comments Invited
We invite you to send any written relevant data, views, or arguments about this proposed AD. Send your comments to an address listed under the ADDRESSES section. Include “Docket No. FAA–2013–0543; Directorate Identifier 2012–NM–202–AD” at the beginning of your comments. We specifically invite comments on the overall regulatory, economic, environmental, and energy aspects of this proposed AD. We will consider all comments received by the closing date and may amend this proposed AD based on those comments.

We will post all comments we receive, without change, to http://www.regulations.gov, including any personal information you provide. We will also post a report summarizing each substantive verbal contact we receive about this proposed AD.

Discussion
The European Aviation Safety Agency (EASA), which is the Technical Agent for the Member States of the European Community, has issued EASA Airworthiness Directive 2012–0223, dated October 23, 2012 (referred to after this as the Mandatory Continuing Airworthiness Information, or “the MCAI”), to correct an unsafe condition for the specified products. The MCAI states:

Airbus identified a batch of special washers, Part Number (P/N) D5725260120000 and P/N D5725664320000, which were incorrectly manufactured and delivered as spares from the supplier between October 2006 and January 2010. As a result of these manufacturing defects, the affected washers differ geometrically from the design specifications. The results of further analysis of Airbus A318, A319, A320 and A321 aeroplanes demonstrate that the affected washers could be seated incorrectly when installed on aeroplanes, which could affect the main landing gear (MLG) retraction jack anchorage fitting bearing installation. This condition, if not detected and corrected, could lead to a local stress concentration which may reduce the fatigue life of the jack fitting, possibly reducing the structural integrity of the affected MLG.

For the reasons described above, this [EASA] AD requires a one-time detailed visual inspection of the left-hand (LH) and right-hand (RH) MLG retraction jack anchorage fitting bearing assemblies to verify that the special washer is seated correctly and, depending on findings, the accomplishment of applicable [related investigative action and] corrective actions.

The related investigative action is a detailed inspection of the jack anchorage fitting for damage, corrosion, cracks or other defects. Corrective actions include replacing the special washer with a new special washer and repairing the jack anchorage fitting if there are signs of damage, corrosion, or other defects. You may obtain further information by examining the MCAI in the AD docket.

Relevant Service Information
• Task 57–26–13–400–001–A, Installation of the Bearing Assembly of the Forward Pintle Pin.
• Task 57–26–13–400–002–A, Installation of the Bearing Assembly of the MLG Actuator Attachment.
• Task 57–26–13–400–004–A, Installation of the Bearing Seals of the MLG Actuator Bearing Assembly.

The actions described in this service information are intended to correct the unsafe condition identified in the MCAI.

FAA’s Determination and Requirements of This Proposed AD
This product has been approved by the aviation authority of another country, and is approved for operation in the United States. Pursuant to our bilateral agreement with the State of Design Authority, we have been notified of the unsafe condition described in the MCAI and service information referenced above. We are proposing this AD because we evaluated all pertinent information and determined an unsafe condition exists and is likely to exist or develop on other products of the same type design

Differences Between This Proposed AD and the MCAI or Service Information
Airbus Service Bulletin A320–57–1169, Revision 01, dated September 18, 2012, specifies to contact the manufacturer for instructions on how to repair certain conditions, but this proposed AD would require repairing those conditions using a method approved by either the Manager, International Branch, ANM–116, Transport Airplane Directorate, FAA; or the EASA (or its delegated agent).

Costs of Compliance
Based on the service information, we estimate that this proposed AD would affect about 851 products of U.S. registry. We also estimate that it would take about 3 work-hours per product to comply with the basic requirements of this proposed AD. The average labor rate is $85 per work-hour. Based on these figures, we estimate the cost of the proposed AD on U.S. operators to be...
$217,005, or $255 per product. In addition, we estimate that any necessary follow-on actions would take about 15 work-hours, for a cost of $1,275 per product. We have received no definitive data that would enable us to provide part cost estimates for the on-condition actions specified in this proposed AD. We have no way of determining the number of products that may need these actions.

**Authority for This Rulemaking**

Title 49 of the United States Code specifies the FAA’s authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. “Subtitle VII: Aviation Programs,” describes in more detail the scope of the Agency’s authority.

We are issuing this rulemaking under the authority described in “Subtitle VII, Part A, Subpart III, Section 44701: General requirements.” Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

**Regulatory Findings**

We determined that this proposed AD would not have federalism implications under Executive Order 13132. This proposed AD would not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify this proposed regulation:

1. Is not a “significant regulatory action” under Executive Order 12866;
2. Is not a “significant rule” under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979);
3. Will not affect intrastate aviation in Alaska; and
4. Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

We prepared a regulatory evaluation of the estimated costs to comply with this proposed AD and placed it in the AD docket.

**List of Subjects in 14 CFR Part 39**

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

**The Proposed Amendment**

Accordingly, under the authority delegated to me by the Administrator, the FAA proposes to amend 14 CFR part 39 as follows:

**PART 39—AIRWORTHINESS DIRECTIVES**

1. The authority citation for part 39 continues to read as follows:

   **Authority:** 49 U.S.C. 106(g), 40113, 44701.

2. The FAA amends §39.13 by adding the following new AD:

   **Airbus:** Docket No. FAA–2013–0543;

   **Directorate Identifier 2012–NM–202–AD.**

   (a) **Comments Due Date**

   We must receive comments by August 15, 2013.

   (b) **Affected ADs**

   None.

   (c) **Applicability**

   This AD applies to the Airbus airplanes listed in paragraphs (c)(1), (c)(2), (c)(3), and (c)(4) of this AD, certificated in any category, all manufacturer serial numbers.


   (d) **Subject**

   Air Transport Association (ATA) of America Code 57, Wings.

   (e) **Reason**

   This AD was prompted by a determination that certain special washers used in retraction jack anchorage fitting bearing installation in the main landing gear (MLG) were incorrectly manufactured. We are issuing this AD to detect and correct installation of incorrectly manufactured special washers, which could lead to a local stress concentration resulting in possible reduction of the fatigue life of the jack fitting, and consequent reduction of the structural integrity of the affected MLG.

   (f) **Compliance**

   You are responsible for having the actions required by this AD performed within the compliance times specified, unless the actions have already been done.

   (g) **Detailed Inspection**

   Within 21,300 flight cycles after August 1, 2006, or within 30 days after the effective date of this AD, whichever occurs later: Do a detailed inspection of the left-hand (LH) and right-hand (RH) MLG retraction jack anchorage fitting bearing assemblies for correct installation, in accordance with the Accomplishment Instructions of Airbus Service Bulletin A320–57–1169, Revision 01, dated September 18, 2012, except as specified in paragraphs (h)(1) and (h)(2) of this AD.

   **Note 1 to paragraph (g) of this AD:** The affected special washers having part numbers (P/N) D572560120000 and P/N D572564320000 were manufactured between October 2006 and January 2010.

   (h) **Related Investigative and Corrective Actions**

   If any special washer is found incorrectly seated during the inspection specified in paragraph (g) of this AD: Before further flight, do all applicable related investigative and corrective actions, in accordance with the Accomplishment Instructions of Airbus Service Bulletin A320–57–1169, Revision 01, dated September 18, 2012, except as specified in paragraph (h)(3) of this AD.

   (i) **Exceptions to Inspections and Service Information**

   (1) Airplanes on which Airbus modification 39730 or Airbus modification 150311 has been embodied in production, or on which Airbus Service Bulletin A320–57–1157 has been embodied in service, do not have to be inspected as required by paragraph (g) of this AD, unless a special washer having P/N D572560120000 or P/N D572564320000 has been installed since the airplane’s first flight, or since modification as specified in Airbus Service Bulletin A320–57–1157, as applicable. A review of airplane maintenance records is acceptable to make this determination if the part numbers of the special washers and modification status can be conclusively determined from that review.

   (2) MLG retraction jack anchorage fitting bearing assemblies on which no special washer replacement has been accomplished after August 1, 2006; and MLG retraction jack anchorage fitting bearing assemblies on which a special washer replacement has been accomplished as specified in Task 57–26–13–400–001–A, Installation of the Bearing Assembly of the Forward Pintle Pin: Task 57–26–13–400–002–A, Installation of the Bearing Assembly of the MLG Actuator Attachment; and Task 57–26–13–400–004–A Installation of the Bearing Seals of the MLG Actuator Bearing Assembly; of Subject 57–26–13, Attachment—Main Landing Gear, of Chapter 57, Wings, of the Airbus A318/A319/A320/A321 Aircraft Maintenance Manual (AMM), Revision 50, dated November 1, 2012; do not have to be inspected as required by paragraph (g) of this AD. A review of airplane maintenance records is acceptable to make this determination if the status can be conclusively determined from that review.

   (3) Where Airbus Service Bulletin A320–57–1169, Revision 01, dated September 18, 2012, specifies to correct Airbus and apply corrective action defined by Airbus: Before further flight, repair the jack anchorage fitting using a method approved by either the Manager, International Branch, ANM–116, Transport Airplane Directorate, FAA; or the European Aviation Safety Agency (EASA) or its delegated agent.
(j) Parts Installation Limitations

As of the effective date of this AD, no person may install, on any airplane, a special washer having P/N D5725260120000 or P/N D5725664320000, unless it is installed in accordance with the Accomplishment Instructions of Airbus Service Bulletin A320–57–1169, Revision 01, dated September 18, 2012; or in accordance with the instructions specified in the tasks identified in paragraphs (j)(1), (j)(2), and (j)(3) of this AD.


(k) Credit for Previous Actions

This paragraph provides credit for actions required by paragraph (g) of this AD, if those actions were performed before the effective date of this AD, using Airbus Service Bulletin A320–57–1169, dated January 10, 2012, which is not incorporated by reference in this AD.

(l) Other FAA AD Provisions

The following provisions also apply to this AD:

(1) Alternative Methods of Compliance (AMOCs): The Manager, International Branch, ANM–116, Transport Airplane Directorate, FAA, has the authority to approve ADs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or local Flight Standards District Office, as appropriate. If sending information directly to the International Branch, send it to ATTN: Sanjay Kalhan, Aerospace Engineer, International Branch, ANM–116, Transport Airplane Directorate, FAA, 1601 Lind Avenue SW., Renton, WA 98057–3356; telephone (425) 227–1405; fax (425) 227–1149. Information may be emailed to: 9-ANM-116-AMOC-REQUESTS@faa.gov. Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the local flight standards district office/ certificate holding district office. The AMOC approval letter must specifically reference this AD.

(2) Airworthy Product: For any requirement in this AD to obtain corrective actions from a manufacturer or other source, use these actions if they are FAA-approved. Corrective actions are considered FAA-approved if they are approved by the State of Design Authority (or their delegated agent). You are required to assure the product is airworthy before it is returned to service.

(m) Related Information


(D) DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39


RIN 2120–AA64

Airworthiness Directives; The Boeing Company Airplanes

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Notice of proposed rulemaking (NPRM).

SUMMARY: We propose to supersede an existing airworthiness directive (AD) that applies to all The Boeing Company Model 737–700, –800, –900, –900ER, –900ER, –900R, –1000, –1000ER, –1000ER, –1000R, –400, and –500 series airplanes. The existing AD currently requires repetitive inspections for discrepancies of each carriage spindle of the outboard mid-flaps; repetitive gap checks of the inboard and outboard carriages of the outboard mid-flaps to detect fractured carriage spindles; measuring to ensure that any new or serviceable carriage spindle meets minimum allowable diameter measurements taken at three locations; repetitive inspections, measurements, and overhaul of the carriage spindles; replacement of any carriage spindle when it has reached its maximum life limit; and corrective actions if necessary. Since we issued that AD, we received a report of failure of both flap carriages. This proposed AD would require reducing the life limit of the carriages, reducing the repetitive interval for certain inspections and gap checks for certain carriages. This proposed AD would also add an option, for certain replacements, of doing an inspection, and related investigative and corrective actions if necessary. We are proposing this AD to detect and correct cracked, corroded, or fractured carriage spindles, which could lead to severe flap asymmetry, and could result in reduced control or loss of controllability of the airplane.

DATES: We must receive comments on this proposed AD by August 15, 2013.

ADDRESSES: You may send comments, using the procedures found in 14 CFR 11.43 and 11.45, by any of the following methods:

• Federal eRulemaking Portal: Go to http://www.regulations.gov. Follow the instructions for submitting comments.

• Fax: 202–482–2251.


• Hand Delivery: Deliver to Mail address above between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

For service information identified in this proposed AD, contact Boeing Commercial Airplanes, Attention: Data & Services Management, P. O. Box 3707, MC 2H–65, Seattle, WA 98124–2207; telephone 206–544–5000, extension 1; fax 206–756–5680; Internet https://www.myboeingfleet.com. You may review copies of the referenced service information at the FAA, Transport Airplane Directorate, 1601 Lind Avenue SW., Renton, WA. For information on the availability of this material at the FAA, call 425–227–1221.

Issued in Renton, Washington, on June 17, 2013.

Jeffrey E. Duven, Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. 2013–15163 Filed 6–28–13; 8:45 am]

BILLING CODE 4910–13–P