

Therefore, this proposed rule is categorically excluded from further NEPA review.

ICE seeks any comments or information that may lead to the discovery of any significant environmental effects from this proposed rule.

List of Subjects in 8 CFR Part 214

Administrative practice and procedure, Aliens, Cultural exchange programs, Employment, Foreign officials, Health professions, Reporting and recordkeeping requirements, Students.

For the reasons discussed in the preamble, DHS proposes to amend Chapter I of Title 8 of the Code of Federal Regulations as follows:

PART 214 — NONIMMIGRANT CLASSES

- 1. The authority citation for part 214 continues to read as follows:

Authority: 8 U.S.C. 1101, 1102, 1103, 1182, 1184, 1186a, 1187, 1221, 1281, 1282, 1301–1305 and 1372; sec. 643, Pub. L. 104–208, 110 Stat. 3009–708; Pub. L. 106–386, 114 Stat. 1477–1480; section 141 of the Compacts of Free Association with the Federated States of Micronesia and the Republic of the Marshall Islands, and with the Government of Palau, 48 U.S.C. 1901 note, and 1931 note, respectively; 48 U.S.C. 1806; 8 CFR part 2.

- 2. In § 214.2 revise paragraph (f)(15)(ii) and paragraph (m)(17)(ii) to read as follows:

§ 214.2 Special requirements for admission, extension, and maintenance of status.

* * * * *

(f) * * *

(15) * * *

(i) * * *

(ii) *Study.*

(A) *F–2 post-secondary/vocational study.*

(1) *Authorized Study at SEVP-Certified Schools.* An F–2 spouse or F–2 child may enroll in less than a full course of study, as defined in 8 CFR 214.2(f)(6)(i)(A)–(D) and 8 CFR 214.2(m)(9)(i)–(iv), in any course of study described in 8 CFR 214.2(f)(6)(i)(A)–(D) or 214.2(m)(9)(i)–(iv) at an SEVP-certified school. Notwithstanding 8 CFR 214.2(f)(6)(i)(B) and 8 CFR 214.2(m)(9)(i), study at an undergraduate college or university or at a community college or junior college is not a full course of study solely because the F–2 nonimmigrant is engaging in a lesser course load to complete a course of study during the current term. An M–2 spouse or M–2 child enrolled in less than a full course of study is not eligible to engage in employment pursuant to paragraph (14) of this subsection.

to engage in employment pursuant to paragraphs (9) and (10) of this subsection.

(2) *Full Course of Study.* Subject to paragraph (f)(15)(ii)(B) and (18), an F–2 spouse and child may engage in a full course of study only by applying for and obtaining a change of status to F–1, M–1 or J–1 nonimmigrant status, as appropriate, before beginning a full course of study. However, an F–2 spouse and child may engage in study that is avocational or recreational in nature, up to and including on a full-time basis.

(B) *F–2 elementary or secondary study.* An F–2 child may engage in full-time study, including any full course of study, in any elementary or secondary school (kindergarten through twelfth grade).

(C) An F–2 spouse violates his or her nonimmigrant status by enrolling in any study except as provided in paragraph (m)(17)(ii)(A) or (B) of this section.

* * * * *

(m) * * *

(17) * * *

(i) * * *

(ii) *Study.*

(A) *M–2 post-secondary/vocational study.*

(1) *Authorized Study at SEVP-Certified Schools.* An M–2 spouse or M–2 child may enroll in less than a full course of study, as defined in 8 CFR 214.2(f)(6)(i)(A)–(D) or 214.2(m)(9)(i)–(v), in any course of study described in 8 CFR 214.2(m)(9)(i)–(v) at an SEVP-certified school. Notwithstanding 8 CFR 214.2(f)(6)(i)(B) and 8 CFR 214.2(m)(9)(i), study at an undergraduate college or university or at a community college or junior college is not a full course of study solely because the M–2 nonimmigrant is engaging in a lesser course load to complete a course of study during the current term. An M–2 spouse or M–2 child enrolled in less than a full course of study is not eligible to engage in employment pursuant to paragraph (14) of this subsection.

(2) *Full Course of Study.* Subject to paragraph (m)(17)(ii)(B), an M–2 spouse and child may engage in a full course of study only by applying for and obtaining a change of status to F–1, M–1, or J–1 status, as appropriate, before beginning a full course of study. However, an M–2 spouse and M–2 child may engage in study that is avocational or recreational in nature, up to and including on a full-time basis.

(B) *M–2 elementary or secondary study.* An M–2 child may engage in full-time study, including any full course of study, in any elementary or secondary

school (kindergarten through twelfth grade).

(C) An M–2 spouse or child violates his or her nonimmigrant status by enrolling in any study except as provided in paragraph (m)(17)(ii)(A) or (B) of this section.

* * * * *

- 3. Revise section 214.3 paragraph (l)(1)(iii) to read as follows:

§ 214.3 Approval of schools for enrollment of F and M nonimmigrants.

(l) * * *

(1) * * *

(i) * * *

(ii) * * *

(iii) School officials may nominate as many DSOs in addition to PDSOs as they determine necessary to adequately provide recommendations to F and/or M students enrolled at the school regarding maintenance of nonimmigrant status and to support timely and complete recordkeeping and reporting to DHS, as required by this section. School officials must not permit a DSO or PDSO nominee access to SEVIS until DHS approves the nomination.

* * * * *

Rand Beers,

Acting Secretary of Homeland Security.

[FR Doc. 2013–27898 Filed 11–20–13; 8:45 am]

BILLING CODE 9111–28–P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA–2013–0997; Directorate Identifier 2013–CE–044–AD]

RIN 2120-AA64

Airworthiness Directives; Slingsby Aviation Ltd. Airplanes

AGENCY: Federal Aviation Administration (FAA), Department of Transportation (DOT).

ACTION: Notice of proposed rulemaking (NPRM).

SUMMARY: We propose to adopt a new airworthiness directive (AD) for Slingsby Aviation Ltd. Model T67M260 airplanes. This proposed AD results from mandatory continuing airworthiness information (MCAI) originated by an aviation authority of another country to identify and correct an unsafe condition on an aviation product. The MCAI describes the unsafe condition as cracked horizontal stabilizer attachment brackets, which could lead to separation of the

horizontal stabilizer and result in loss of control. We are issuing this proposed AD to require actions to address the unsafe condition on these products.

DATES: We must receive comments on this proposed AD by January 6, 2014.

ADDRESSES: You may send comments by any of the following methods:

- *Federal eRulemaking Portal:* Go to <http://www.regulations.gov>. Follow the instructions for submitting comments.

- *Fax:* (202) 493-2251.

- *Mail:* U.S. Department of

Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue SE., Washington, DC 20590.

• *Hand Delivery:* U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue SE., Washington, DC 20590, between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

For service information identified in this proposed AD, contact Slingsby Advanced Composites, Ings Lane, Kirbymoorside, York, YO62 6EZ, United Kingdom, telephone: +44 (0) 1751 432474; fax +44 (0) 1751 433016, Internet: www.marshall-slingsby.com. You may review this referenced service information at the FAA, Small Airplane Directorate, 901 Locust, Kansas City, Missouri 64106. For information on the availability of this material at the FAA, call (816) 329-4148.

Examining the AD Docket

You may examine the AD docket on the Internet at <http://www.regulations.gov> by searching for and locating it in Docket No. FAA-2013-0997; or in person at the Docket Management Facility between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this proposed AD, the regulatory evaluation, any comments received, and other information. The street address for the Docket Office (telephone (800) 647-5527) is in the **ADDRESSES** section. Comments will be available in the AD docket shortly after receipt.

FOR FURTHER INFORMATION CONTACT:

Mike Kiesov, Aerospace Engineer, FAA, Small Airplane Directorate, 901 Locust, Room 301, Kansas City, Missouri 64106; telephone: (816) 329-4144; fax: (816) 329-4090; email: mike.kiesov@faa.gov.

SUPPLEMENTARY INFORMATION:

Comments Invited

We invite you to send any written relevant data, views, or arguments about this proposed AD. Send your comments to an address listed under the

ADDRESSES section. Include "Docket No. FAA-2013-0997; Directorate Identifier 2013-CE-044-AD" at the beginning of your comments. We specifically invite comments on the overall regulatory, economic, environmental, and energy aspects of this proposed AD. We will consider all comments received by the closing date and may amend this proposed AD because of those comments.

We will post all comments we receive, without change, to <http://regulations.gov>, including any personal information you provide. We will also post a report summarizing each substantive verbal contact we receive about this proposed AD.

Discussion

The European Aviation Safety Agency (EASA), which is the Technical Agent for the Member States of the European Community, has issued AD No. 2012-0169, dated August 31, 2012 (referred to after this as "the MCAI"), to correct an unsafe condition for the specified products. The MCAI states:

Several cases have been reported of cracked horizontal stabiliser attachment brackets on Slingsby T67 aeroplanes.

This condition, if not detected and corrected, could lead to separation of the horizontal stabiliser and consequent loss of control of the aeroplane.

Prompted by these reports, Slingsby issued Service Bulletin (SB) 179 to provide instructions for repetitive inspections. The CAA UK, the State of Design authority at the time, issued AD 001-12-2002, which was later superseded by AD G-2005-0004 (EASA approval 2005-564) to require repetitive inspections and, depending on findings, replacement of the affected brackets.

Since that AD was issued, Slingsby published SB 179 issue 4, which removed the Model T67M260-T3A from the Applicability (all aeroplanes of this Model are confirmed to have been scrapped) and clarified that replacement of the affected aluminum brackets with titanium brackets (Slingsby Modification M988A or B) constitutes terminating action for the repetitive inspections.

For the reasons described above, this AD retains the requirements of CAA UK AD G-2005-0004, which is superseded, removes the Model T67M260-T3A from the Applicability and confirms that installing titanium brackets constitutes terminating action for the repetitive inspection requirements of this AD.

You may examine the MCAI on the Internet at <http://www.regulations.gov> by searching for and locating it in Docket No. FAA-2013-0997.

Relevant Service Information

Slingsby Advanced Composites Ltd. has issued Service Bulletin No. 179, Issue 4, dated March 15, 2007. The

actions described in this service information are intended to correct the unsafe condition identified in the MCAI.

FAA's Determination and Requirements of the Proposed AD

This product has been approved by the aviation authority of another country, and is approved for operation in the United States. Pursuant to our bilateral agreement with this State of Design Authority, they have notified us of the unsafe condition described in the MCAI and service information referenced above. We are proposing this AD because we evaluated all information and determined the unsafe condition exists and is likely to exist or develop on other products of the same type design.

Costs of Compliance

We estimate that this proposed AD will affect 11 products of U.S. registry. We also estimate that it would take about 2 work-hours per product to comply with the basic inspection of the aluminum horizontal stabilizer attachment brackets requirement of this proposed AD. The average labor rate is \$85 per work-hour.

Based on these figures, we estimate the cost of the proposed AD on U.S. operators to be \$1,870, or \$170 per product.

In addition, we estimate that any necessary follow-on actions would take about 8 work-hours and require parts costing \$7,250 (for all four titanium horizontal stabilizer attachment brackets), for a cost of \$7,930 per product, or parts costing \$9,557 (for all four aluminum horizontal stabilizer attachment brackets), for a cost of \$10,237. We have no way of determining the number of products that need these actions.

Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. "Subtitle VII: Aviation Programs," describes in more detail the scope of the Agency's authority.

We are issuing this rulemaking under the authority described in "Subtitle VII, Part A, Subpart III, section 44701: General requirements." Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority.

because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

Regulatory Findings

We determined that this proposed AD would not have federalism implications under Executive Order 13132. This proposed AD would not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify this proposed regulation:

- (1) Is not a "significant regulatory action" under Executive Order 12866,
- (2) Is not a "significant rule" under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979),

(3) Will not affect intrastate aviation in Alaska, and

(4) Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

The Proposed Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA proposes to amend 14 CFR part 39 as follows:

PART 39—AIRWORTHINESS DIRECTIVES

- 1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

- 2. The FAA amends § 39.13 by adding the following new AD:

Slingsby Aviation Ltd.: Docket No. FAA-2013-0997; Directorate Identifier 2013-CE-044-AD.

(a) Comments Due Date

We must receive comments by January 6, 2014.

(b) Affected ADs

None.

(c) Applicability

This AD applies to Slingsby Aviation Ltd. Model T67M260 airplanes, all serial numbers, certificated in any category.

(d) Subject

Air Transport Association of America (ATA) Code 55: Stabilizers.

(e) Reason

This AD was prompted by mandatory continuing airworthiness information (MCAI) originated by an aviation authority of another country to identify and correct an unsafe condition on an aviation product. The MCAI describes the unsafe condition as cracked horizontal stabilizer attachment brackets. We are issuing this AD to prevent separation of the horizontal stabilizer, which could result in loss of control.

(f) Actions and Compliance

Unless already done, do the actions specified in paragraphs (f)(1) through (f)(4) of this AD:

(1) Within the next 150 hours time-in-service (TIS) after the effective date of this AD or at the next annual inspection after the effective date of this AD, whichever occurs later, and repetitively thereafter at intervals not to exceed 150 hours TIS, inspect the aluminum horizontal stabilizer attachment brackets for cracks. Do the inspections following the ACTION instructions in Slingsby Advanced Composites Ltd. Service Bulletin S.B. No: 179, Issue 4, dated March 15, 2007.

(2) If, during any inspection required in paragraph (f)(1) of this AD, any cracks are found, before further flight, replace the cracked bracket with a serviceable part. Do the replacement following the ACTION instructions in Slingsby Advanced Composites Ltd. Service Bulletin S.B. No: 179, Issue 4, dated March 15, 2007. If a serviceable aluminum horizontal stabilizer attachment bracket is used as a replacement part, repetitively inspect as specified in paragraph (f)(1) of this AD.

(3) To terminate the repetitive inspections required in paragraph (f)(1) of this AD, all four aluminum horizontal stabilizer attachment brackets must be replaced with titanium horizontal stabilizer attachment brackets.

(4) After installing titanium horizontal stabilizer attachment brackets, installing aluminum horizontal stabilizer attachment brackets are prohibited.

(g) Other FAA AD Provisions

The following provisions also apply to this AD:

(1) *Alternative Methods of Compliance (AMOCs):* The Manager, Standards Office, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. Send information to ATTN: Mike Kiesov, Aerospace Engineer, FAA, Small Airplane Directorate, 901 Locust, Room 301, Kansas City, Missouri 64106; telephone: (816) 329-4144; fax: (816) 329-4090; email: mike.kiesov@faa.gov. Before using any approved AMOC on any airplane to which the AMOC applies, notify your appropriate principal inspector (PI) in the FAA Flight Standards District Office (FSDO), or lacking a PI, your local FSDO.

(2) *Airworthy Product:* For any requirement in this AD to obtain corrective actions from a manufacturer or other source, use these actions if they are FAA-approved. Corrective actions are considered FAA-approved if they are approved by the State of Design Authority (or their delegated agent). You are required

to assure the product is airworthy before it is returned to service.

(3) *Reporting Requirements:* For any reporting requirement in this AD, a federal agency may not conduct or sponsor, and a person is not required to respond to, nor shall a person be subject to a penalty for failure to comply with a collection of information subject to the requirements of the Paperwork Reduction Act unless that collection of information displays a current valid OMB Control Number. The OMB Control Number for this information collection is 2120-0056. Public reporting for this collection of information is estimated to be approximately 5 minutes per response, including the time for reviewing instructions, completing and reviewing the collection of information. All responses to this collection of information are mandatory. Comments concerning the accuracy of this burden and suggestions for reducing the burden should be directed to the FAA at: 800 Independence Ave. SW., Washington, DC 20591, Attn: Information Collection Clearance Officer, AES-200.

(h) Related Information

Refer to European Aviation Safety Agency (EASA) AD No. 2012-0169, dated August 31, 2012, for related information. You may examine the MCAI on the Internet at <http://www.regulations.gov> by searching for and locating it in Docket No. FAA-2013-0997. For service information related to this AD, contact Slingsby Advanced Composites, Ings Lane, Kirbymoorside, York, YO62 6EZ, United Kingdom, telephone: +44 (0) 1751 432474; fax +44 (0) 1751 433016, Internet: www.marshall-slingsby.com. You may review this referenced service information at the FAA, Small Airplane Directorate, 901 Locust, Kansas City, Missouri 64106. For information on the availability of this material at the FAA, call (816) 329-4148.

Issued in Kansas City, Missouri, on November 15, 2013.

Earl Lawrence,

Manager, Small Airplane Directorate, Aircraft Certification Service.

[FR Doc. 2013-27919 Filed 11-20-13; 8:45 am]

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 71

[Docket No. FAA-2013-0921; Airspace Docket No. 13-AAL-4]

Proposed Modification of Class E Airspace; Sitka, AK

AGENCY: Federal Aviation Administration (FAA), Department of Transportation (DOT).

ACTION: Notice of proposed rulemaking (NPRM).

SUMMARY: This action proposes to modify Class E airspace at Sitka, AK, to